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1 **Effects of Isothermal Initial Stratification Strength on Vorticity Dynamics for**  
2 **Single-Mode Compressible Rayleigh-Taylor Instability**

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8 **Abstract**

9 The effects of isothermal initial stratification on the dynamics of the vorticity for single-mode Rayleigh-  
10 Taylor instability (RTI) are examined using two-dimensional fully compressible wavelet-based direct numer-  
11 ical simulations. The simulations model low Atwood number ( $A = 0.04$ ) RTI development for four different  
12 initial stratification strengths, corresponding to Mach numbers from 0.3 (weakly stratified) to 1.2 (strongly  
13 stratified), and for three different Reynolds numbers, from 25,500 to 102,000. Here, the Mach number is  
14 based on the Atwood-independent gravity wave speed and characterizes the strength of the initial stratifi-  
15 cation. All simulations use adaptive wavelet-based mesh refinement to achieve very fine spatial resolutions  
16 at relatively low computational cost. For all stratifications, the RTI bubble and spike go through the ex-  
17 ponential growth regime, followed by a slowing of the RTI evolution. For the weakest stratification, this  
18 slow-down is then followed by a re-acceleration, while for stronger stratifications the suppression of RTI  
19 growth continues. Bubble and spike asymmetries are observed for weak stratifications, with bubble and  
20 spike growth rates becoming increasingly similar as the stratification strength increases. For the range of  
21 cases studied, there is relatively little effect of Reynolds number on bubble and spike heights, although the  
22 formation of secondary vortices becomes more pronounced as Reynolds number increases. The underlying  
23 dynamics are analyzed in detail through an examination of the vorticity transport equation, revealing that  
24 incompressible baroclinicity drives RTI growth for small and moderate stratifications, but increasingly leads  
25 to the suppression of vorticity production and RTI growth for stronger stratifications. These variations in  
26 baroclinicity are used to explain the suppression of RTI growth for strong stratifications, as well as the  
27 anomalous asymmetry in bubble and spike growth rates for weak stratifications.

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## 28 I. INTRODUCTION

29 Rayleigh-Taylor instability (RTI) is formed at the interface of two fluids with different densities  
30 when an accelerative force is applied across the interface in the direction of the less dense fluid  
31 [1, 2]. Such a scenario arises in a number of practical engineering and physics problems, including  
32 inertial confinement fusion (ICF) [3, 4], supernova ignition fronts [5–9], X-ray bursts [10], and  
33 various topics in geophysics [11–13], to name just a few examples. In many of these problems,  
34 as well as in many experiments [13, 16–19], the Atwood number (namely, the density ratio of the  
35 two fluids normalized to take values between 0 and 1) is small, the background or initial state is  
36 stratified, and the Reynolds number is large, resulting in compressible dynamics driven by relatively  
37 small density differences over a wide range of length and time scales. In the present study, two  
38 dimensional (2D) fully compressible wavelet-based direct numerical simulations (DNS) are used  
39 to examine, from a dynamical standpoint, the evolution of low Atwood number RTI for different  
40 isothermal initial stratification strengths and Reynolds numbers.

41 The present focus on initial stratification strength is intended to reveal the effects of flow  
42 compressibility on RTI evolution. Both flow and fluid compressibility may affect RTI growth; the  
43 former is related to the thermodynamic state and the stratification of background density and  
44 pressure fields, while the latter is related to the equation of state and differences in the specific  
45 heat ratio between the two fluids [46]. Flow compressibility is associated with quantities that  
46 are independent of fluid properties, for example the velocity or thermodynamic state, while fluid  
47 compressibility relates to material properties that can only be changed by changing the fluid itself.  
48 Gauthier [15] refers to these two types of compressibility as “static” and “dynamic” compressibility,  
49 respectively. In the present context, the strength of the initial stratification is given by a Mach  
50 number characteristic of flow compressibility, namely  $M = \sqrt{g\lambda}/a$ , where  $g$  is the gravitational  
51 acceleration,  $\lambda$  is the wavelength of the initial perturbation used to generate the RTI, and  $a$  is  
52 the sound speed. The Atwood number does not appear in this definition of  $M$ , since the Atwood  
53 number is most relevant to the change in fluid properties at the interface between the two fluids  
54 and is thus more directly associated with fluid compressibility.

55 Until relatively recently, many DNS studies of RTI used incompressible, low-Mach number,  
56 Boussinesq, or anelastic approximations to reduce the computational cost, often yielding valuable  
57 physical insights (see, e.g., [20–24, 46]). However, the present study is one of a growing number of  
58 fully compressible DNS analyses of RTI growth and characteristics. Lafay *et al.* [25] examined RTI  
59 growth in the linear regime for different compressibility strengths (addressing both flow and fluid

60 compressibility), and Gauthier [26] examined RTI growth into the nonlinear regime for two different  
61 stratification strengths. More recently, Reckinger *et al.* [27] examined single-mode, 2D RTI growth  
62 rates for a range of stratification strengths, and Gauthier [15] performed a comprehensive study  
63 of the dynamics of multi-mode, three-dimensional (3D) RTI for a relatively strongly stratified  
64 case. Both of these more recent studies employed variable-resolution numerical methods to achieve  
65 high Reynolds numbers within the context of fully compressible DNS; Reckinger *et al.* [27] used  
66 the parallel adaptive wavelet collocation method (PAWCM) [28] and Gauthier [15] used an auto-  
67 adaptive multi-domain Chebyshev-Fourier method [29]. Using currently available computational  
68 resources, these and other adaptive techniques are unavoidable when performing fully compressible  
69 DNS at high Reynolds numbers. The present study correspondingly employs PAWCM to study  
70 the effects of flow compressibility and Reynolds number, including high Reynolds numbers, on RTI  
71 growth and dynamics.

72 Based in large part on observations from these prior fully compressible DNS studies, the general  
73 effects of compressibility on RTI growth are now relatively well understood. Despite some initial  
74 ambiguity regarding the specific impacts of compressibility (dating back, at least, to the studies  
75 by Bernstein & Book [30] and Baker [31]), Livescu [46] used a linear analysis of the Navier-Stokes  
76 equations to show that, for isothermal background stratification, flow compressibility is associated  
77 with a reduction in the rate of RTI growth, while fluid compressibility is associated with an increase  
78 in the growth rate, as compared to the corresponding incompressible case. A number of studies  
79 have confirmed these results, particularly with respect to the suppression of RTI growth by flow  
80 compressibility [15, 25, 27, 32–34]. In particular, as the stratification strength of the background  
81 density field increases, an increasing suppression of RTI growth has been observed. Reckinger *et al.*  
82 [27] further found that there are asymmetries in the locations and speeds of upward propagating  
83 low density fluid (i.e., “bubbles”) and downward propagating high density fluid (i.e., “spikes”),  
84 even at relatively small Atwood number, that may be different than in the incompressible limit. It  
85 was also shown by Reckinger *et al.* [27] that drag and potential flow models are unable to predict  
86 the suppression of RTI growth for strong stratifications.

87 Compared to the effects of flow compressibility, Reynolds number effects on RTI growth have  
88 received somewhat less attention (although the Péclet number is more directly related to the  
89 balance between convective and diffusive effects in RTI, the Schmidt and Prandtl numbers are  
90 taken as unity, or close to unity, in nearly all prior simulation studies, resulting in a correspondence  
91 between the Péclet and Reynolds numbers). Wei & Livescu [34] used the incompressible variable-  
92 density form of the Navier-Stokes equations to show that, at early non-dimensional times  $t\sqrt{Ag/\lambda}$ ,

93 where  $t$  is time and  $A$  is the Atwood number, RTI growth rates are larger for smaller Reynolds  
94 numbers due to diffusive effects. At long times, however, RTI growth rates were found to be  
95 greater for larger Reynolds numbers. The crossover in growth rates between low and high Reynolds  
96 numbers was found to occur at non-dimensional times of roughly 3 – 4, corresponding to the end  
97 of the potential flow growth stage of the RTI. There are indications, however, that the RTI re-  
98 accelerates at later times [35] and may, in fact, grow quadratically at sufficiently high Reynolds  
99 numbers [34], contrary to the “terminal velocity” assumption in previous studies. In this case,  
100 single-mode RTI may represent an upper bound for the multi-mode case. The single-mode growth  
101 rate was also found to become independent of Reynolds number at sufficiently large Reynolds  
102 numbers. Using fully compressible DNS, Gauthier [15] similarly found that smaller Reynolds  
103 numbers are associated with faster early growth rates of the turbulent mixing layer produced by  
104 the RTI. At later times, growth rates for higher Reynolds numbers are similar to, or exceed, those of  
105 lower Reynolds numbers. These results were, however, obtained for a single stratification strength,  
106 and it remains to be seen how these Reynolds number effects depend on stratification strength,  
107 if at all. It should be noted that these Reynolds number effects are likely associated with the  
108 observation by Dimotakis [36] that, when the Reynolds number is sufficiently high, small-scale  
109 turbulent features develop beyond the mixing transition and further increases in the Reynolds  
110 number do not yield significant changes to the turbulence characteristics. In the Rayleigh-Taylor  
111 literature, this has been explored, for example, by Cook *et al.* [37].

112 In order to understand compressibility and Reynolds number effects in more detail, several au-  
113 thors have examined the dynamics of the vorticity during RTI evolution, generally finding that  
114 changes in the baroclinic torque are responsible for changes in RTI growth rates. Lafay *et al.*  
115 [25] examined the linear regime and found that vorticity production decreases as the stratifica-  
116 tion strength increases. More recently, Schneider & Gauthier [38] performed a systematic study  
117 of vorticity during RTI growth using 3D multimode simulations that employ the Boussinesq ap-  
118 proximation. This study showed that there is an increase in the strength of baroclinic torque  
119 production with time, although the contribution to the overall dynamics is dwarfed by the effects  
120 of nonlinear vortex stretching. Gauthier [15, 26] was the first to examine vorticity using fully  
121 compressible DNS, and showed the importance of baroclinic torque in producing vorticity during  
122 RTI growth for a single strongly stratified case. However, changes to the relative magnitudes of  
123 the various terms in the vorticity transport equation for different stratification strengths are still  
124 not completely understood in the fully compressible case.

125 Despite the improved understanding of compressibility and Reynolds number effects provided

126 by the recent, primarily computational, studies noted above, a number of outstanding questions  
 127 remain, and the present study is specifically focused on addressing the following: (i) How does the  
 128 behavior of low Atwood number RTI depend on both initial stratification strength and Reynolds  
 129 number?; (ii) What are the dynamical causes of the observed RTI phenomena?; and (iii) How  
 130 do the dynamics (specifically, the vorticity dynamics) depend on initial stratification strength?  
 131 The first question is motivated by the studies of Lafay *et al.* [25] and Wei & Livescu [34]; the  
 132 former studied the effects of compressibility, but within the linear regime and for only one Reynolds  
 133 number, while the latter studied a range of Reynolds numbers, but using an incompressible variable-  
 134 density formulation of the Navier-Stokes equations that precluded the study of compressibility  
 135 effects. The second and third questions are motivated primarily by the studies of Reckinger *et al.*  
 136 [27], Schneider & Gauthier [38], and Gauthier [15]. The first of these studies observed bubble-spike  
 137 asymmetries but not their dynamical causes, while the second and third studies both performed  
 138 extensive analyses of the vorticity dynamics, but using the Boussinesq approximation (i.e., not a  
 139 fully compressible study) and for only one stratification strength, respectively. It should also be  
 140 noted that several prior studies [13, 39, 40] have examined incompressible RTI in the presence  
 141 of stable background stratification, and here we examine fully compressible RTI under similar  
 142 circumstances, with the notable distinction that the present stratification is vertically asymmetric.

143 In the present paper, DNS are performed at low Atwood numbers (0.04 here, as compared  
 144 to 0.1-0.7 in [27]) for different Reynolds numbers and different strengths of initial hydrostatic  
 145 stratification, corresponding to Mach numbers between 0.3 (weak stratification) and 1.2 (strong  
 146 stratification) [41, 42, 46]. The DNS are performed using adaptive mesh refinement based on  
 147 PAWCM, as described, validated, and implemented for RTI by Reckinger *et al.* [27, 28]. This  
 148 method allows high spatial resolution to be used where it is needed (e.g., where density and  
 149 velocity gradients are large), while reducing the total number of computational collocation points.  
 150 The present focus on low Atwood numbers is motivated primarily by the observations of quadratic  
 151 high Reynolds number single-mode RTI growth in regimes with similarly low Atwood number in  
 152 the study by Wei & Livescu [34]. In order to understand the dynamical causes of the observed  
 153 results, the various terms in the vorticity transport equation are examined as functions of time and  
 154 stratification strength.

155 It should be noted that several simplifications are made here to allow the underlying physics  
 156 to be more easily understood. In particular, complex interactions of multiple wavelengths are  
 157 eliminated by applying only single-mode initial perturbations to the unstable interface between the  
 158 two fluids with differing densities. Moreover, in the classical incompressible case, where the density

159 of both fluids is constant, RTI growth eventually leads to a re-acceleration of the bubble and spike  
 160 tips, finally resulting in chaotic dynamics and development. The compressible case is, however,  
 161 more complicated due to spatial and temporal variations in the background density, pressure, and  
 162 temperature fields. The effects of changing any of these fields are largely unknown, and thus only  
 163 isothermal initial stratifications are studied here to eliminate thermal effects, since the initial state  
 164 is already in thermal equilibrium. Future work will explore the effects of multi-modal perturbations  
 165 and different stratification types. Finally, the present simulations and analysis are performed in  
 166 2D in order to enable the examination of several different stratification strengths and Reynolds  
 167 numbers. Each such simulation is computationally expensive and performing a similarly expansive  
 168 study in 3D remains the focus of future research, due primarily to the need for substantially more  
 169 computational resources. The primary disadvantage of the present 2D approach is the resulting  
 170 lack of nonlinear vortex stretching in the vorticity dynamics, although the absence of this effect  
 171 does have the benefit of more clearly revealing the effects of baroclinicity on the dynamics.

172 The rest of this paper is organized as follows. The next section discusses the problem setup, in-  
 173 cluding the governing equations and initialization of the RTI. Section III provides a brief discussion  
 174 of how the wavelet-based adaptive method (i.e., PAWCM) was used to complete the simulations.  
 175 In Section IV, the paper goes in depth into the results of this study, looking at the effects of strati-  
 176 fication strength and Reynolds number on RTI growth. In Section V, the dynamics of the vorticity  
 177 for fully compressible RTI are outlined and examined. Finally, a summary and conclusions are  
 178 presented in Section VI.

## 179 II. DESCRIPTION OF THE PHYSICAL PROBLEM

180 In the present study, RTI occurs through the initial placement of a heavier fluid, denoted by  
 181 index ‘2’ with molar mass  $W_2$ , above a lighter fluid, denoted by index ‘1’ with molar mass  $W_1$ ,  
 182 in the presence of a gravitational accelerative force. The addition of a perturbation leads to the  
 183 onset of the RTI, and the heavier fluid begins to fall into the lighter fluid in a spike-like formation,  
 184 while the lighter fluid rises into the heavier fluid in a bubble-like formation. For the present  
 185 low Atwood number cases, “bubbles” are defined as upward-traveling low density features, while  
 186 spikes are downward-traveling high density features. In the following, the fully compressible fluid  
 187 flow equations solved by the DNS are outlined, followed by a description of the initial isothermal  
 188 hydrostatic stratifications of different strengths (as characterized by a static Mach number). It  
 189 should be noted that the equations solved are identical to those used in the study by Reckinger *et*

190 *al.* [27], but are repeated here since they are the starting point for the study of vorticity dynamics  
 191 in Section V.

## 192 A. Governing Equations

193 The numerical simulations solve the fully compressible Navier-Stokes equations for two miscible  
 194 fluids given by [41]

$$\frac{\partial \rho}{\partial t} + \frac{\partial(\rho u_j)}{\partial x_j} = 0, \quad (1)$$

$$\frac{\partial(\rho u_i)}{\partial t} + \frac{\partial(\rho u_i u_j)}{\partial x_j} = -\frac{\partial p}{\partial x_i} + \rho g_i + \frac{\partial \tau_{ij}}{\partial x_j}, \quad (2)$$

$$\frac{\partial(\rho e)}{\partial t} + \frac{\partial(\rho e u_j)}{\partial x_j} = -\frac{\partial(p u_i)}{\partial x_i} + \rho u_i g_i + \frac{\partial(\tau_{ij} u_i)}{\partial x_j} - \frac{\partial q_j}{\partial x_j} + \frac{\partial[T(c_p)_l s_{jl}]}{\partial x_j}, \quad (3)$$

$$\frac{\partial(\rho Y_i)}{\partial t} + \frac{\partial(\rho Y_i u_j)}{\partial x_j} = \frac{\partial s_{ji}}{\partial x_j}, \quad (4)$$

195 where  $\rho$  is the density,  $u_i$  is the velocity in the  $x_i$  direction,  $p$  is the pressure,  $g_i$  is the gravitational  
 196 acceleration,  $\tau_{ij}$  is the viscous stress tensor,  $e$  is the specific total energy,  $q_i$  is the heat flux,  $T$  is  
 197 the temperature,  $(c_p)_l$  is the specific heat capacity at constant pressure for fluid  $l$ ,  $s_{ji}$  is the mass  
 198 flux for fluid  $i$  in the  $x_j$  direction, and  $Y_i$  is the mass fraction for the  $i^{\text{th}}$  fluid. Note that, for a  
 199 two-fluid system,  $Y_2 = 1 - Y_1$ , and so Eq. (4) is only solved in the present simulations for  $i = 2$   
 200 (i.e., the heavier fluid). The pressure and caloric ideal gas laws are assumed to hold, so that the  
 201 pressure and specific total energy can be expressed as

$$p = \rho R T, \quad (5)$$

$$e = \frac{1}{2} u_i u_i + c_v T, \quad (6)$$

202 where  $R$  is the mixture gas constant defined in terms of the universal gas constant  $\mathcal{R}$  and the molar  
 203 mass of each fluid,  $W_i$ , as

$$R = Y_i R_i = \mathcal{R} \frac{Y_i}{W_i}. \quad (7)$$

204 In the above expression, the species gas constant is defined as  $R_i \equiv \mathcal{R}/W_i$ . The mixture specific  
 205 heat at constant volume,  $c_v$ , appearing in Eq. (6) is similarly defined as

$$c_v = (c_v)_i Y_i, \quad (8)$$



206 where the specific heats at constant pressure and volume are related by  $(c_p)_i = (c_v)_i + R_i$  and their  
 207 mixture values by  $c_p = c_v + R$ . The specific heats at constant volume are assumed constant and  
 208 the same for the two fluids, so that the mixture specific heat at constant pressure varies with the  
 209 flow due to the different molar masses of the two fluids.

210 The viscous stress  $\tau_{ij}$  in Eqs. (2) and (3) is assumed to be Newtonian and is given by

$$\tau_{ij} = \mu \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} - \frac{2}{3} \frac{\partial u_k}{\partial x_k} \delta_{ij} \right) = 2\mu S'_{ij}, \quad (9)$$

211 where  $S'_{ij} = S_{ij} - S_{kk}\delta_{ij}/3$  is the deviatoric strain rate and the dynamic viscosity is given by  
 212  $\mu = \rho\nu$ , with the kinematic viscosity  $\nu$  assumed to be constant (i.e., temperature independent  
 213 and the same for both fluids) such that spatial and temporal variations in  $\mu$  are due entirely to  
 214 variations in  $\rho$ . The strain rate tensor,  $S_{ij}$ , is given by

$$S_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right). \quad (10)$$

215 The heat flux in Eq. (3) is written as

$$q_j = -k \frac{\partial T}{\partial x_j}, \quad (11)$$

216 where  $k$  is the thermal conductivity, and the species mass flux in Eqs. (3) and (4) is defined as

$$s_{ji} = \rho D \frac{\partial Y_i}{\partial x_j}, \quad (12)$$

217 where  $D$  is the mass diffusivity. For the range of parameters considered here, the baro-diffusion  
 218 term is small in the mass flux, and Soret and Dufour effects are neglected in the mass and heat  
 219 fluxes, respectively. Both  $k$  and  $D$ , like the kinematic viscosity  $\nu$ , are assumed to be constant and  
 220 temperature-independent, and both Prandtl and Schmidt numbers are unity, resulting in an exact  
 221 correspondence between the Reynolds and Péclet numbers.

222 The majority of fluid properties are taken to be the same between the two fluids for simplicity.  
 223 This includes the kinematic viscosity,  $\nu$ , the heat conduction coefficient,  $k$ , and the mass diffusion  
 224 coefficient,  $D$ . It should be noted that effects due to bulk viscosity and non-equilibrium thermo-  
 225 dynamics are neglected in the simulations. Investigating these effects is beyond the scope of the  
 226 present study, although Sagert *et al.* [43] and Lai *et al.* [44] have recently made progress in this  
 227 direction.

228 The system of equations given by Eqs. (1)-(12) is solved using the PAWCM numerical approach,  
 229 which is described in Section III, for an RTI with a physical setup as outlined in the following  
 230 section.

## 231 B. Initialization of Rayleigh Taylor Instability

232 The RTI problem is initialized in the DNS by imposing a perturbation on a stratified isothermal  
 233 background state that is in hydrostatic equilibrium. The gravitational acceleration is assumed to  
 234 be in the negative  $x_1$  direction, such that  $g_i = -g\delta_{i1}$ , where  $g$  is the magnitude of the gravitational  
 235 acceleration. The resulting density,  $\rho(x_1, x_2, t)$ , and pressure,  $p(x_1, x_2, t)$ , fields at  $t = 0$  can be  
 236 expressed as

$$\rho(x_1, x_2, 0) = \rho_0(x_1) + \rho'(x_1, x_2, 0), \quad (13)$$

$$p(x_1, x_2, 0) = p_0(x_1) + p'(x_1, x_2, 0), \quad (14)$$

237 where  $\rho_0$  and  $p_0$  are hydrostatic initial background states and  $\rho'(x_1, x_2, 0)$  and  $p'(x_1, x_2, 0)$  represent  
 238 the initial perturbations to the background states.

239 Assuming an isothermal background state at temperature  $T_0$ , the background density and pres-  
 240 sure fields for fluid  $\alpha$  (where  $\alpha = [1, 2]$  and summation over Greek indices is not implied) are given  
 241 by

$$\rho_{0\alpha}(x_1) = \frac{p_I}{R_\alpha T_0} \exp\left(-\frac{gx_1}{R_\alpha T_0}\right), \quad (15)$$

$$p_{0\alpha}(x_1) = p_I \exp\left(-\frac{gx_1}{R_\alpha T_0}\right), \quad (16)$$

242 where the initial interface between the two fluids lies at  $x_1 = 0$ ,  $p_I$  is the interfacial pressure, and  
 243  $R_\alpha = \mathcal{R}/W_\alpha$  is the gas constant based on the molar mass of fluid  $\alpha$ . The heavier fluid ( $\alpha = 2$ )  
 244 is initially located above the interface for  $x_1 > 0$  and the lighter fluid ( $\alpha = 1$ ) is initially located  
 245 below the interface for  $x_1 < 0$ . A corresponding interfacial density is given using the ideal gas law  
 246 as  $\rho_I = p_I/(R_I T_0)$  where  $R_I = \mathcal{R}[(W_1 + W_2)/2]^{-1}$ .

247 In each of the cases examined here, the kinematic viscosity  $\nu = \mu/\rho$ , which is constant and the  
 248 same in both fluids, is set using the Reynolds number,  $Re$ , defined as

$$Re \equiv \sqrt{\frac{g\lambda^3}{\nu^2}} \quad \Rightarrow \quad \nu = \sqrt{\frac{g\lambda^3}{Re^2}}, \quad (17)$$

249 where  $\lambda$  is the wavelength of the applied perturbation. The non-dimensional Atwood number,  $A$ ,  
 250 is defined as

$$A \equiv \frac{W_2 - W_1}{W_2 + W_1}. \quad (18)$$

251 Note that in the present study,  $W_2 > W_1$  in order to generate RTI. It should be noted that the  
 252 definition of  $Re$  in Eq. (17) does not include  $A$ , despite its appearance in the definition of the  
 253 perturbation Reynolds number,  $Re_p = Re[A/(1 + A)]^{1/2}$  in previous studies (e.g., [34]) of low-  
 254 Atwood RTI. Here,  $A$  is not included in  $Re$  in order to ensure a consistent non-dimensionalization  
 255 of the vorticity dynamics in Section V based only on  $g$ ,  $\lambda$ , and the interfacial density  $\rho_I$ . The  
 256 definition of  $Re$  in Eq. (17) is also consistent with the prior PAWCM study by Reckinger *et al.*  
 257 [27].

258 The degree of flow compressibility defined by the thermodynamic conditions enters the RTI  
 259 problem by affecting both the background stratification and the further development of dilata-  
 260 tional (non-zero velocity divergence) effects [41, 46]. While dilatational effects and their acoustic  
 261 manifestations are usually characterized by the Mach number denoting the ratio between veloc-  
 262 ity and sound speed, together with dilatational to solenoidal kinetic energy ratios, stratification  
 263 strength can also be recast as a Mach number. This can be done by re-expressing  $gx_1/(R_\alpha T_0)$  in  
 264 Eqs. (15) and (16), as described below.

265 In the present study, the relevant incompressible limit is found by simultaneously increasing  
 266 the background pressure and temperature to cause an increase in the speed of sound such that  
 267 the density remains unaffected. This incompressible limit is also easily obtained in practice by  
 268 uniformly heating a fixed volume of fluid. This results in the definition of an isothermal Mach  
 269 number based on the ratio of the Atwood-independent gravity wave speed,  $\sqrt{g\lambda}$ , and the isothermal  
 270 speed of sound,  $a_0 = \sqrt{p_I/\rho_I}$  [32, 46]. The resulting Mach number,  $M$ , is then given by

$$M = \sqrt{\frac{\rho_I g \lambda}{p_I}} \quad \Rightarrow \quad M^2 = \frac{g \lambda}{R_I T_0}. \quad (19)$$

271 It should be noted that  $M$  is equivalent to stratification strength parameters used in prior studies  
 272 of flow (or “static”) compressibility [15, 23, 45], and that larger values of  $M$  indicate stronger initial  
 273 stratification. The Atwood number,  $A$ , is not included in the definition of  $M$  since the present Mach  
 274 number is intended to be characteristic of the initial background stratification, which is independent  
 275 of  $A$ . Similar Mach number definitions have also been used in prior studies of compressible RTI

276 (e.g., [46]). A Mach number characterizing fluid (or “dynamic”) compressibility, by contrast, would  
 277 be expected to include  $A$ .

278 Normalizing  $\rho_{0\alpha}$  in Eq. (15) and  $p_{0\alpha}$  in Eq. (16) by  $\rho_I$ ,  $g$ , and  $\lambda$ , the non-dimensional background  
 279 states can be rewritten as

$$\rho_{0\alpha}^*(x_1^*) = \frac{R_I}{R_\alpha} \exp\left(-M^2 \frac{R_I}{R_\alpha} x_1^*\right), \quad (20)$$

$$p_{0\alpha}^*(x_1^*) = \frac{1}{M^2} \exp\left(-M^2 \frac{R_I}{R_\alpha} x_1^*\right), \quad (21)$$

280 where the characteristic pressure is given as  $\rho_I g \lambda$  and  $x_1^* \equiv x_1/\lambda$  is a normalized distance variable.  
 281 It can be shown that the ratio  $R_I/R_\alpha$  can be written in terms of the Atwood number  $A$  as

$$\frac{R_I}{R_\alpha} = \frac{2W_\alpha}{W_1 + W_2} = 1 + (-1)^\alpha A \text{ for } \alpha = 1, 2. \quad (22)$$

282 Since  $\alpha = 1$  corresponds to the lighter fluid for which  $x_1^* < 0$  initially and  $\alpha = 2$  corresponds to the  
 283 heavier fluid for which  $x_1^* > 0$ , the non-dimensional background states  $\rho_0^*$  and  $p_0^*$  can be written in  
 284 final form as

$$\rho_0^*(x_1^*) = (1 \pm A) \exp[-M^2(1 \pm A)x_1^*], \quad (23)$$

$$p_0^*(x_1^*) = \frac{1}{M^2} \exp[-M^2(1 \pm A)x_1^*], \quad (24)$$

285 where  $\rho_0^* = \rho_0/\rho_I$ ,  $p_0^* = p_0/(\rho_I g \lambda)$ , with  $(1 - A)$  for  $x_1^* < 0$  (i.e., the lighter fluid) and  $(1 + A)$   
 286 for  $x_1^* > 0$  (i.e., the heavier fluid). The resulting initial background stratifications are shown for a  
 287 variety of Mach numbers in Figure 1, where the size of the density difference at  $x_1^* = 0$  is determined  
 288 by the value of  $A$  ( $A = 0.04$  in the present study).

289 Following the procedure extensively outlined by Reckinger *et al.* [27], a single-mode velocity  
 290 perturbation was applied at  $t = 0$  to initialize the RTI. Although they are not perfect represen-  
 291 tations of multi-mode engineering problems found in ICF and other practical applications, the  
 292 present single-mode simulations can nevertheless be used to gain insights into compressibility-  
 293 driven physics and dynamics. As shown by Reckinger *et al.* [27], single-mode simulations can  
 294 expose any numerical directional bias in the code, which is generally hidden in multi-mode simu-  
 295 lations. As a result, single-mode simulations allow the opportunity to ensure that the simulations  
 296 are completely resolved from the initial state through to late times, and also allow simple checks  
 297 for symmetry and the introduction of extraneous perturbation modes throughout the simulation.

298 In addition, the results of Wei & Livescu [34] show that single-mode RTI may represent the upper  
 299 bound for the multi-mode growth rate at low Atwood numbers, when the Reynolds number is  
 300 sufficiently large.

### 301 **III. DETAILS OF THE DIRECT NUMERICAL SIMULATIONS**

302 Due to the spatial localization of the developing region, the RTI problem lends itself naturally to  
 303 state-of-the-art adaptive grid numerical methods. In particular, to effectively capture the instability  
 304 evolution, very long domains are needed to ensure that late-time growth is captured, but very small  
 305 grid spacing is required to fully resolve the high gradients at the interface of the instability. For a  
 306 static computational grid with fixed cell size, this results in a very dense grid and incredibly high  
 307 computational costs. During the majority of the simulation, however, very fine grid resolutions  
 308 far away from the interface are unnecessary and, as a result, high grid compression ratios can  
 309 be achieved through the use of adaptive grid approaches. A method that has proven effective at  
 310 achieving high compression ratios is the Parallel Adaptive Wavelet Collocation Method (PAWCM)  
 311 [27, 28], which is the method that is applied here.

#### 312 **A. Wavelet-Based Grid Adaptation**

313 The PAWCM numerical approach has been applied previously to the simulation of compressible  
 314 RTI by Reckinger *et al.* [27], where validation and details of the numerical method are exhaustively  
 315 outlined. These details are repeated only briefly here, and the reader is referred to [27] for additional  
 316 information.

317 Fundamentally, PAWCM uses the natural properties of the wavelet transform to locate areas of  
 318 steep gradients and to provide direct control over the grid cell size used to resolve the gradients.  
 319 Essentially, through PAWCM, a flow field variable is transformed into wavelet space, resulting in  
 320 wavelet basis functions and coefficients that are localized in both wave and physical spaces. From  
 321 there, the coefficients are passed through a thresholding filter where all of the coefficients with  
 322 magnitudes above the parameter  $\varepsilon$  are kept, and any of those below  $\varepsilon$  are set to zero. The resulting  
 323 thresholded decomposition can thus be written for a generic variable  $f$  as

$$f_{\geq}(x) = \sum_k c_k^0 \phi_k^0(x) + \sum_{j=0}^{\infty} \sum_{\alpha=1}^{2^n-1} \sum_l d_l^{\alpha,j} \psi_l^{\alpha,j}(x), \quad (25)$$

$$|d_l^{\alpha,j}| \geq \varepsilon \|f\|$$

324 where  $\phi_k$  are scaling functions on the coarsest level,  $c_k$  are the corresponding coarse-level wavelet  
 325 coefficients,  $\psi_l$  are the scaling interpolating functions on any arbitrary level,  $d_l$  are the coefficients  
 326 to which the thresholding is applied,  $l$  and  $k$  represent physical grid points, and  $\alpha$  and  $j$  represent  
 327 the wavelet family and level of resolution, respectively [47, 48]. The effect of setting any one of the  
 328 coefficients  $d_l$  to zero is the removal of a grid point at that level of resolution. These coefficients  
 329 take on large values for large gradients, and small values in relatively uniform regions. The effective  
 330 resolution is set by a base grid size and the limit put on  $j$  (referred to as  $j_{max}$  herein). This results  
 331 in the error being  $\mathcal{O}(\varepsilon)$  and the resolution in a single direction being  $p \cdot 2^{(j_{max}-1)}$ , where  $p$  is the  
 332 base resolution [47–49].

333 As outlined in Reckinger *et al.* [27], PAWCM has been implemented in a way that enables  
 334 it to work with finite difference approaches to solving governing equations such as those in Eqs.  
 335 (1)-(4). In solving these equations, fourth-order central differences have been applied spatially, and  
 336 a third-order total variation diminishing explicit Runge-Kutta scheme has been applied in time.  
 337 The PAWCM algorithm is highly parallelized, having successfully run on up to 5,000 cores, and is  
 338 able to perform arbitrary domain decompositions using the Zoltan library. It has a tree-like data  
 339 structure for easy MPI communications, as well as direct error control. As a result, the additional  
 340 computational overhead introduced by the wavelet methodology is offset by the capability to use  
 341 many processors and to achieve grid compression ratios greater than 90% [47–49].

342 Substantial discussion was provided in Reckinger *et al.* [27] regarding the flow variables on  
 343 which to adapt the grid in the DNS. Since the wavelet method is so flexible, it is possible to adapt  
 344 the grid on any flow field variable that is calculable and of interest. In the present study, adaptation  
 345 for the initial time steps was performed using the vorticity, the norm of the strain rate tensor, and  
 346 the gradient of the species mass fraction  $Y_2$ , in addition to the velocity and mass fraction fields.  
 347 This approach allowed the RTI to develop with sufficient accuracy prior to further refining the grid  
 348 on more complex flow variables at later times to reflect the increasing complexity of the flow. In  
 349 particular, at late times in the present study, adaptation was performed using the baroclinic torque  
 350 to ensure that this dynamically important term was fully resolved for the analysis of the vorticity  
 351 dynamics. Additional details on grid convergence and resolution can be found in [27].

## 352 B. Simulation Setup

353 In the present study, PAWCM is used to solve the governing equations outlined in Section II A  
 354 for the background and initial conditions described in Section II B. The simulations have been

355 carried out in 2D and the total domain size was  $16\lambda$  in the  $x_1$  direction and  $\lambda$  in the  $x_2$  direction,  
 356 where  $\lambda$  corresponds to the wavelength of the applied perturbation. The maximum effective grid  
 357 resolution resulting from the adaptive wavelet approach was  $\Delta x^* = 2.4 \times 10^{-4}$ , where  $\Delta x^* = \Delta x/\lambda$   
 358 and  $\Delta x$  is the grid cell size. This results in a maximum of 4,096 grid cells in the  $x_2$  direction, which  
 359 occurs primarily near  $x_1^* = 0$  where the RTI develops. Although there is a potential maximum of  
 360 65,536 points in the  $x_1$  direction, the adaptive wavelet method only provides high resolution near  
 361 the RTI and, thus, each simulation includes far fewer points along the  $x_1$  direction.

362 The Atwood number studied was 0.04, and the Mach numbers used were 0.3 (nearly incompress-  
 363 ible), 0.6, 0.9, and 1.2. The Reynolds numbers,  $Re$ , investigated were 25,500, 51,000, and 102,000  
 364 (corresponding to perturbation Reynolds numbers,  $Re_p$ , of 5,000, 10,000, and 20,000, respectively),  
 365 giving a total of twelve simulations performed in the present study (i.e., four different values of  $M$ ,  
 366 and three values of  $Re$  for each  $M$ ). The highest Reynolds number is of particular interest because  
 367 it has been shown to be the minimum perturbation Reynolds number necessary to reach the chaotic  
 368 growth regime for the incompressible limit (i.e.,  $M \rightarrow 0$ ) of this particular case [34]. Each of the  
 369 simulations were performed up to a non-dimensional time of  $t^* = t/\sqrt{\lambda/g} = 20$ , corresponding to  
 370 the time at which the bubble and spike had reached heights of roughly  $\lambda$  (or  $x_1^* = \pm 1$ ) for the  
 371  $M = 0.3$  case.

372 Boundaries in the  $x_2$  direction are taken to be periodic. In the  $x_1$  direction, at the top and  
 373 bottom of the domain, shear-free slip boundary conditions were implemented with numerical dif-  
 374 fusion buffer zones immediately before each boundary interior to the domain. The purpose of  
 375 these “open” boundary conditions is to essentially mimic an infinite domain and to ensure that  
 376 both the background stratification is preserved and that none of the shocks introduced by the RTI  
 377 initialization are reflected back into the domain. In particular, the buffer zones ensure that any  
 378 shockwaves are dissipated prior to reaching the boundaries [27].

379 As discussed in [27], some artificial thickening of the interface at  $x_1^* = 0$  and  $t^* = 0$  can be  
 380 beneficial since the thicker interface can act as a buffer layer to absorb other numerical errors.  
 381 In general, however, thicker interfaces have the potential to introduce asymmetries in the initial  
 382 conditions which propagate as undesirable longer-time asymmetries during RTI growth. Based on  
 383 these two competing considerations, the number of points across the interface was chosen to be 16,  
 384 to both minimize the asymmetry and to gain some measure of beneficial buffering effects. Finally,  
 385 it was found that higher resolutions led to better initial conditions. At a level of  $j_{max} = 7$ , it was  
 386 found that the asymmetry drops below machine precision, and thus this level of resolution was  
 387 deemed sufficient for the present simulations.

#### 388 IV. RAYLEIGH TAYLOR INSTABILITY GROWTH AND CHARACTERISTICS

389 The PAWCM-enabled simulations performed here are designed to allow examination of stratifi-  
390 cation strength (as parameterized by  $M$ ) and Reynolds number (as parameterized by  $Re$ ) effects on  
391 RTI growth and characteristics. In the following, these two effects are investigated with a primary  
392 focus on the heights and velocities of bubbles and spikes formed during the RTI development. Here  
393 the “height” is denoted  $h$  and refers to the absolute value of the respective distances from  $x_1 = 0$  of  
394 the bubble and spike “tips” in the  $x_1$  direction. The bubble and spike tips correspond to the 99%  
395 and 1% mass fraction values, respectively. Bubble and spike velocities, denoted  $u_h$ , are computed  
396 from the time derivatives of the bubble and spike heights. An analysis of the dynamics underlying  
397 the observed bubble and spike behaviors is outlined in Section V.

##### 398 A. Effects of Stratification Strength

399 Figure 2 shows RTI growth as a function of time for each of the four stratification strengths,  
400 where  $Re = 102,000$  in all cases. For each case, bubbles and spikes form soon after initializing  
401 the simulation and the RTI grows as  $t$  increases. Small-scale features in each case become increas-  
402 ingly pronounced as the RTI evolves, and secondary vortices are most prominent for the weakest  
403 stratification (i.e.,  $M = 0.3$ ). The corresponding bubble and spike growths decrease as the stratifi-  
404 cation strength increases; for the strongest stratification (i.e.,  $M = 1.2$ ), the RTI growth is halted  
405 relatively early in its evolution.

406 Consistent with the fields in Figure 2, Figure 3(a) shows that the suppression of RTI growth  
407 compared to the incompressible (i.e.,  $M \rightarrow 0$ ) case from Wei & Livescu [34] occurs for all strati-  
408 fications considered. For the two strongest stratifications (i.e.,  $M = 0.9$  and  $1.2$ ), the bubble and  
409 spike each reach maximum heights before  $t^* = 4$  and stop growing.

410 The dependence of RTI growth on stratification strength can be investigated further by con-  
411 sidering time series of the bubble and spike tip velocities, as shown in Figure 3(b). This figure  
412 indicates that bubble and spike velocities for the larger Mach numbers all trend towards zero,  
413 indicative of the complete suppression of RTI for strong stratifications. For  $M = 0.3$ , however,  
414 there is a re-acceleration of the spike tip shortly after  $t^* = 15$ .

415 In addition to these changes in the bubble and spike heights with varying stratification strength,  
416 Figure 3(a) also shows that spikes reach consistently greater heights than bubbles for all  $M$ . This  
417 asymmetry, particularly for low  $M$ , is not present in the purely incompressible case of Wei &



418 Livescu [34], where it was found that for the low Atwood number case of 0.04, bubble and spike  
 419 heights were close until after the re-acceleration regime. As indicated by Figure 3(b), the velocities  
 420 at the tips of the spikes are consistently larger than those at the tips of the bubbles, although the  
 421 difference between these velocities becomes significant only for the  $M = 0.3$  case after  $t^* > 10$ .

422 It should be noted that full suppression of RTI growth for all but the lowest value of  $M$  cannot  
 423 be predicted based solely on considerations of the potential energy of the system. This is shown in  
 424 Figure 3(b), where only the lowest value of  $M$  reaches a plateau near the velocity predicted from  
 425 either drag [50, 51] or potential flow [52] models (namely,  $u_h/\sqrt{g\lambda} \approx 0.063$ ).

426 Based on the DNS results for  $M = 0.3$  to 1.2, the primary observations are that larger strati-  
 427 fications are associated with decreasing bubble and spike growth rates, resulting in a suppression  
 428 of the RTI for all but the smallest value of  $M$  studied here, and that smaller stratifications are  
 429 associated with more asymmetric bubble and spike growth rates. This amounts to an anomalous  
 430 asymmetry at low stratifications (i.e.  $M = 0.3$ ), since both zero and large  $M$  limits are more  
 431 symmetrical. The results concerning the suppression of the instability are in general agreement  
 432 with those from prior studies [15, 25, 27, 33, 34]. In particular, the suppression of the instability  
 433 begins at slightly later times as  $M$  increases, consistent with results from, for example, Reckinger  
 434 *et al.* [27]. To better understand the dynamics leading to RTI suppression and the development  
 435 of bubble and spike asymmetries, an analysis of the underlying vorticity dynamics is performed in  
 436 Section V.

## 437 B. Effects of Reynolds Number

438 Figure 4 shows RTI growth for the weakest (i.e.,  $M = 0.3$ ) and strongest (i.e.,  $M = 1.2$ )  
 439 stratifications for Reynolds numbers  $Re = 25,500$ ,  $51,000$ , and  $102,000$ . There is little qualitative  
 440 dependence of the bubble and spike heights on  $Re$ , indicating that these large-scale characteristics  
 441 of RTI growth are already in an asymptotic limit for  $Re = 25,500$ . This is consistent with the  
 442 results from Wei & Livescu [34], where it was found that there is little difference in the RTI growth  
 443 rates before the onset of the very late chaotic development for values of  $Re$  above roughly 7,500.

444 Despite the relative similarity of the large-scale structure for the three values of  $Re$  examined  
 445 here, however, there is substantial dependence of small-scale structure on  $Re$ . In particular, Figure  
 446 4 shows that an increasing amount of small-scale detail emerges as  $Re$  increases, corresponding to  
 447 the occurrence of viscous dissipation at increasingly smaller scales. This increase in scale range  
 448 with increasing  $Re$  results in the formation of secondary vortices for  $M = 0.3$ . Even though there

449 is also increasing small scale structure for  $M = 1.2$  with increasing  $Re$ , the formation of secondary  
450 vortices is less pronounced for this higher stratification due to the overall suppression in the RTI  
451 growth.

452 From a quantitative perspective, Figure 5 shows bubble and spike heights and velocities for each  
453 of the four values of  $M$  examined in the present study. For the bubble and spike heights shown in  
454 Figure 5(a), there is little or no dependence on  $Re$  for any stratification strength. However, for the  
455 velocities in Figure 5(b), there is a clear trend towards faster initial accelerations as  $Re$  increases.  
456 The bubbles and spikes also reach larger maximum velocities as  $Re$  increases. However, at very  
457 early times in the evolution for each  $M$ , during diffusive growth, bubble and spike velocities are  
458 largest for small  $Re$ , eventually crossing over in each case at  $t^* \approx 5$  such that the higher  $Re$  cases  
459 have greater velocities at later times. This result is consistent with the crossover in speeds observed  
460 by Wei & Livescu [34] and, to a somewhat lesser extent, by Gauthier [15].

461 These trends are consistent for all stratification strengths, although the differences with  $Re$   
462 become more pronounced as  $M$  increases. For example, the peak bubble and spike velocities for  
463  $M = 1.2$  are reached at roughly  $t^* = 12$  when  $Re = 102,000$  and at roughly  $t^* = 14$  when  
464  $Re = 25,500$ . After reaching the peak values, however, the bubble and spike velocities become  
465 substantially less dependent on  $Re$ . For the case with smallest  $M$ , the results approach the nearly  
466 incompressible limit (i.e.,  $M \rightarrow 0$ ) where, as shown by Wei & Livescu [34], no dependence on  $Re$  is  
467 observed above  $Re \approx 1,500$  during the times examined here (before the onset of late time chaotic  
468 development regime).

469 Taken together, these results indicate that, for the values of  $Re$  examined here, there is little  
470 dependence of the global RTI growth on  $Re$  during the later stages of the instability at higher  
471 stratifications and through the early re-acceleration stage for  $M = 0.3$ . However, the early time  
472 evolution, small scale structure, and the appearance of secondary vortices are all substantially  
473 affected by  $Re$ . Given the increasing effect of  $Re$  with increasing  $M$ , it may be the case that  $Re$   
474 effects become increasingly pronounced for even stronger stratifications than the  $M = 1.2$  case  
475 examined here; exploring such more strongly stratified scenarios is left as a direction for future  
476 research.

477 **V. VORTICITY DYNAMICS FOR COMPRESSIBLE RAYLEIGH TAYLOR INSTABIL-**  
 478 **ITY**

479 Properties and dynamics of the vorticity vector,  $\omega_i = \epsilon_{ijk} \partial u_k / \partial x_j$ , where  $\epsilon_{ijk}$  is the alternat-  
 480 ing tensor, have been widely studied to understand flow behavior in a variety of contexts. For  
 481 compressible flows more specifically, vorticity has been studied in shock-driven [53–55], reacting  
 482 [56, 57], and various types of buoyant [58] flows, revealing the dynamical importance of variable  
 483 density effects such as dilatation and baroclinic torque. In the case of RTI, however, only Gauthier  
 484 [15] has examined vorticity dynamics in the fully compressible regime, and for only one value of  
 485 the initial stratification strength.

486 In the following sections, properties of the vorticity during RTI growth are outlined as a func-  
 487 tion of stratification strength, and terms in the non-dimensional compressible vorticity transport  
 488 equation are subsequently examined to understand the underlying dynamics. The role of baro-  
 489 clinic torque, in particular, in the suppression of RTI growth for strong stratifications and in the  
 490 formation of bubble and spike asymmetries for weak stratifications is outlined. It should be noted  
 491 that the importance of baroclinic torque in RTI growth is not new or surprising and has been  
 492 highlighted in several previous studies [15, 25, 26, 34, 38]. The primary contribution of the current  
 493 work is in explaining how the baroclinic torque varies with initial stratification strength, as well as  
 494 how RTI suppression and asymmetry arise from a dynamical perspective.

495 **A. Vorticity Evolution for Compressible RTI**

496 In the 2D simulations,  $\omega_3$  is the only nonzero component of the vorticity, and Figure 6 shows  
 497 the temporal evolution of the non-dimensional vorticity  $\omega_3^* = \omega_3 \sqrt{\lambda/g}$  for each of the stratification  
 498 strengths. In each case, the vorticity field initially develops as a vortex pair with generally positive  
 499 vorticity for  $x_2^* < 0.5$  and negative vorticity for  $x_2^* > 0.5$ . These initial vortex pairs evolve by  
 500 moving downwards slowly in the domain, while the Kelvin-Helmholtz instability on the sides of the  
 501 bubbles and spikes sheds further vortex pairs. The overall spatial extent of vorticity production  
 502 is greatest for weak stratification (i.e.,  $M = 0.3$ ), with “fronts” of non-zero vorticity magnitude  
 503 that propagate upwards and downwards in an analogous way to the propagation of bubbles and  
 504 spikes, respectively, as shown in Figure 2. The vorticity evolution at  $M = 0.3$  is reminiscent of the  
 505 overall picture in the incompressible (i.e.,  $M \rightarrow 0$ ) case, with induced vortical velocity supporting  
 506 the instability growth and leading to re-acceleration and late time chaotic development. However,

507 at higher values of  $M$ , no additional vortex pairs are generated. The overall magnitude of the  
 508 vorticity is also shown in Figure 6 to decrease with increasing  $M$ .

509 The overall  $M$ -dependence of the vorticity magnitude is also explored in Figure 7 using the  
 510 vorticity averaged over the half domain, denoted  $\bar{\bar{\omega}}_3$ , where the half-domain averaging operator is  
 511 defined for an arbitrary quantity  $f$  as

$$\bar{\bar{f}}(t) = \frac{2}{\lambda} \int_0^{\lambda/2} \left[ \frac{1}{2\lambda} \int_{-\lambda}^{\lambda} f(x_1, x_2, t) dx_1 \right] dx_2. \quad (26)$$

512 Figure 7 shows that  $\bar{\bar{\omega}}_3$  generally increases at early times at a rate that is larger with decreasing  
 513 stratification. After the initial growth of  $\bar{\bar{\omega}}_3$  shown in Figure 7, the average vorticity decreases with  
 514 time for all but the weakest stratification (i.e.,  $M = 0.3$ ). This result mirrors the suppression of  
 515 RTI growth for all but the weakest stratification, seen in Figure 3.

## 516 B. Non-Dimensional Compressible Vorticity Transport Equation

517 The dynamics governing the evolution of the vorticity in compressible RTI can be understood  
 518 from the non-dimensional vorticity transport equation, which reveals the explicit dependence of the  
 519 dynamics on  $A$ ,  $M$ , and  $Re$ . A similar equation was derived using the Boussinesq approximation by  
 520 Schneider & Gauthier [38], although any explicit dependence on the initial stratification strength  
 521 was omitted in the derivation. Here, the non-dimensional transport equation is derived for the fully  
 522 compressible case, permitting the explicit identification of dependencies on stratification strength  
 523  $M$ .

524 By taking the curl of the momentum equation in Eq. (2), the transport equation for the 3D  
 525 vorticity vector is obtained for a variable density, variable viscosity compressible flow as

$$\frac{D\omega_i}{Dt} = \omega_j S_{ij} - \omega_i S_{kk} - \epsilon_{ijk} \frac{\partial v}{\partial x_j} \frac{\partial p}{\partial x_k} + \epsilon_{ijk} \frac{\partial}{\partial x_j} \left[ v \frac{\partial (2\mu S'_{kl})}{\partial x_l} \right], \quad (27)$$

526 where  $D/Dt \equiv \partial/\partial t + u_i \partial/\partial x_i$  is the Lagrangian derivative and  $v \equiv 1/\rho$  is the specific volume,  
 527 which is used here instead of  $\rho$  to simplify the derivation. The first term on the right-hand side of  
 528 Eq. (27) represents vortex stretching, the second term represents dilatation, which is zero in the  
 529 incompressible limit where  $S_{kk} = 0$ , the third term is the baroclinic torque, and the last term is  
 530 viscous diffusion, where the viscous stress tensor  $\tau_{kl}$  has been expressed in terms of the deviatoric  
 531 strain rate tensor  $S'_{kl}$  [see Eq. (9)]. It should be noted that in Section VC we will examine the  
 532 vorticity evolution in 2D simulations for which the vortex stretching term vanishes exactly.

533 The last term in Eq. (27), representing viscous diffusion, can be separated into an essentially  
 534 incompressible term that is present regardless of whether viscosity,  $\mu$ , is spatially and temporally  
 535 varying, and into a term that is only present when  $\mu$  is non-constant. In the present simulations,  
 536  $\mu = \nu\rho$ , where  $\nu$  is a constant given in terms of problem parameters as in Eq. (17), and  $\rho$  is the  
 537 spatially and temporally varying density. Expansion of the diffusion term in Eq. (27) then gives  
 538 the vorticity transport equation for a variable density, variable viscosity flow as

$$\frac{D\omega_i}{Dt} = \omega_j S_{ij} + \nu \frac{\partial^2 \omega_i}{\partial x_j \partial x_j} - \omega_i S_{kk} - \epsilon_{ijk} \frac{\partial v}{\partial x_j} \frac{\partial p}{\partial x_k} - 2\nu \epsilon_{ijk} \frac{\partial}{\partial x_j} \left( \frac{S'_{kl}}{v} \frac{\partial v}{\partial x_l} \right). \quad (28)$$

539 The first two terms on the right-hand side of this equation are present even in constant density,  
 540 constant viscosity flows, while the last three terms are only nonzero when  $v$  (and, by extension,  
 541 the density) is non-constant.

542 Using the characteristic time scale  $\sqrt{\lambda/g}$  to define the non-dimensional vorticity  $\omega_i^* \equiv \omega_i \sqrt{\lambda/g}$ ,  
 543 and using  $\rho_I g \lambda$  as the characteristic pressure, Eq. (28) can be written in non-dimensional form as

$$\frac{D\omega_i^*}{Dt^*} = \omega_j^* S_{ij}^* + \frac{1}{Re} \frac{\partial^2 \omega_i^*}{\partial x_j^* \partial x_j^*} - \omega_i^* S_{kk}^* - \epsilon_{ijk} \frac{\partial v^*}{\partial x_j^*} \frac{\partial p^*}{\partial x_k^*} - \frac{2}{Re} \epsilon_{ijk} \frac{\partial}{\partial x_j^*} \left[ S_{kl}^* \frac{\partial (\ln v^*)}{\partial x_l^*} \right]. \quad (29)$$

544 Based on the above equation, both diffusive terms scale in an identical way with  $Re$ . It should  
 545 be noted, however, that the stratification strength  $M$  does not appear explicitly in Eq. (29),  
 546 although it is present implicitly in the baroclinic torque term [i.e., the fourth term on the right-  
 547 hand side of Eq. (29)]. To reveal this dependence, the baroclinic torque can be rewritten by defining  
 548 new perturbation variables  $v'^*$  and  $p'^*$  that express  $v^*$  and  $p^*$  relative to their respective  $A = 0$   
 549 background stratifications as

$$v'^*(\mathbf{x}^*, t^*) \equiv v^*(\mathbf{x}^*, t^*) - [v_M^*(x_1^*) - 1], \quad (30)$$

$$p'^*(\mathbf{x}^*, t^*) \equiv p^*(\mathbf{x}^*, t^*) - \left[ p_M^*(x_1^*) + x_1^* - \frac{1}{M^2} \right], \quad (31)$$

550 where  $\mathbf{x}^* = [x_1^*, x_2^*, x_3^*]$ ,  $v_M^* \equiv 1/\rho_M^*$ , and  $\rho_M^*$  and  $p_M^*$  correspond to the  $A = 0$  profiles of  $\rho_0^*$  and  
 551  $p_0^*$  from Eqs. (23) and (24), respectively. The  $A = 0$  profiles are used for normalization purposes  
 552 to avoid discontinuities in the derivatives of the background profiles that arise when  $A$  is nonzero  
 553 (particularly for the first derivative of  $p_0^*$ ). The resulting  $A = 0$  profiles are, nevertheless, not  
 554 substantially different than the  $A = 0.04$  profiles (see Figure 1) and serve the purpose of explicitly  
 555 revealing the dependence of the baroclinic torque on  $M$ .

556 The decompositions in Eqs. (30) and (31) are designed to yield  $v'^* = v^*$  and  $p'^* = p^*$  in the

557 limit as  $M \rightarrow 0$ , as well as  $\overline{\partial v'^*/\partial x_i^*} = \partial v^*/\partial x_i^*$  and  $\overline{\partial p'^*/\partial x_i^*} = \partial p^*/\partial x_i^*$  in the same limit. The  
 558 baroclinic torque term in Eq. (29) depends only on gradients of  $v^*$  and  $p^*$ , and the equivalency of  
 559 the perturbation and total gradients can be shown for  $M \rightarrow 0$  as

$$\frac{\partial v'^*}{\partial x_i^*} = \frac{\partial v^*}{\partial x_i} - M^2 v_M^* \delta_{i1} \quad \Rightarrow \quad \frac{\partial v'^*}{\partial x_i^*} = \frac{\partial v^*}{\partial x_i} \text{ as } M \rightarrow 0, \quad (32)$$

$$\frac{\partial p'^*}{\partial x_i^*} = \frac{\partial p^*}{\partial x_i} + (M^2 p_M^* - 1) \delta_{i1} \quad \Rightarrow \quad \frac{\partial p'^*}{\partial x_i^*} = \frac{\partial p^*}{\partial x_i} \text{ as } M \rightarrow 0, \quad (33)$$

560 where  $v_M^* \rightarrow 1$  and  $M^2 p_M^* \rightarrow 1$  as  $M \rightarrow 0$ . As  $M$  becomes large and background stratification  
 561 becomes increasingly strong, the magnitude of  $\partial v'^*/\partial x_i^*$  becomes increasingly small and  $\partial p'^*/\partial x_i^*$   
 562 approaches the background stratification everywhere, as shown in Figure 8.

563 Using Eqs. (32) and (33), it can be shown that the baroclinic torque on the right-hand side of  
 564 Eq. (29) can be written as

$$-\epsilon_{ijk} \frac{\partial v^*}{\partial x_j^*} \frac{\partial p^*}{\partial x_k^*} = -\epsilon_{ijk} \frac{\partial v'^*}{\partial x_j^*} \frac{\partial p'^*}{\partial x_k^*} + (M^2 p_M^* - 1) \epsilon_{ij1} \frac{\partial v'^*}{\partial x_j^*} + (M^2 v_M^*) \epsilon_{ij1} \frac{\partial p'^*}{\partial x_j^*}, \quad (34)$$

565 where it is assumed that  $v_M^*$  and  $p_M^*$  depend only on  $x_1^*$ . The first term on the right in Eq. (34)  
 566 represents the baroclinic torque that is independent of the initial background stratification, and  
 567 this is the only remaining term in the limit as  $M \rightarrow 0$ . The second and third terms represent the  
 568 baroclinic torques associated with the initial stratified background pressure and specific volume  
 569 fields, respectively. It should be noted that the present analysis is specific to the isothermal forms  
 570 for  $v_M^*$  and  $p_M^*$  obtained from Eqs. (32) and (33), and that the scaling may differ for different initial  
 571 background conditions (e.g., isentropic or isobaric conditions).

572 After substituting Eq. (34) into Eq. (29), the non-dimensional 3D vorticity transport equation  
 573 is obtained for a compressible flow with initial background stratification as

$$\begin{aligned} \frac{D\omega_i^*}{Dt^*} = & \omega_j^* S_{ij}^* + \frac{1}{Re} \frac{\partial^2 \omega_i^*}{\partial x_j^* \partial x_j^*} - \omega_i^* S_{kk}^* - \epsilon_{ijk} \frac{\partial v'^*}{\partial x_j^*} \frac{\partial p'^*}{\partial x_k^*} \\ & + (M^2 p_M^* - 1) \epsilon_{ij1} \frac{\partial v'^*}{\partial x_j^*} + (M^2 v_M^*) \epsilon_{ij1} \frac{\partial p'^*}{\partial x_j^*} - \frac{2}{Re} \epsilon_{ijk} \frac{\partial}{\partial x_j^*} \left[ S_{kl}^* \frac{\partial (\ln v^*)}{\partial x_l^*} \right], \end{aligned} \quad (35)$$

574 where, once more, the first four terms are present even in the limit as  $M \rightarrow 0$  and the fifth  
 575 and sixth terms are only significant for nonzero  $M$ . The corresponding transport equation for the

576 vorticity magnitude  $\omega^* \equiv (\omega_i^* \omega_i^*)^{1/2}$  is given by

$$\begin{aligned}
\frac{D\omega^*}{Dt^*} = & \underbrace{\widehat{\omega}_i^* \omega_j^* S_{ij}^*}_{\mathcal{T}_1^*} + \underbrace{\frac{\widehat{\omega}_i^*}{Re} \frac{\partial^2 \omega_i^*}{\partial x_j^* \partial x_j^*}}_{\mathcal{T}_2^*} - \underbrace{\omega^* S_{kk}^*}_{\mathcal{T}_3^*} - \underbrace{\widehat{\omega}_i^* \epsilon_{ijk} \frac{\partial v^*}{\partial x_j^*} \frac{\partial p^*}{\partial x_k^*}}_{\mathcal{T}_4^*} \\
& + \underbrace{(M^2 p_M^* - 1) \widehat{\omega}_i^* \epsilon_{ij1} \frac{\partial v^*}{\partial x_j^*}}_{\mathcal{T}_5^*} + \underbrace{(M^2 v_M^*) \widehat{\omega}_i^* \epsilon_{ij1} \frac{\partial p^*}{\partial x_j^*}}_{\mathcal{T}_6^*} - \underbrace{\frac{2\widehat{\omega}_i^*}{Re} \epsilon_{ijk} \frac{\partial}{\partial x_j^*} \left[ S_{kl}^* \frac{\partial(\ln v^*)}{\partial x_l^*} \right]}_{\mathcal{T}_7^*}.
\end{aligned} \tag{36}$$

577 where  $\widehat{\omega}_i^* \equiv \omega_i^*/\omega^*$  is the vorticity unit vector (where the magnitude of  $\widehat{\omega}_i^*$  is unity by definition).  
578 This expression is valid for any  $M$ ,  $A$ , and  $Re$  provided that  $v_M^*$  and  $p_M^*$  are given by Eqs. (32) and  
579 (33) and that  $\nu$  is constant. In the above expression,  $\mathcal{T}_1^*$  represents production and destruction of  $\omega^*$   
580 due to vortex stretching,  $\mathcal{T}_2^*$  represents diffusion of vorticity by viscosity,  $\mathcal{T}_3^*$  represents dilatational  
581 effects,  $\mathcal{T}_4^*$  represents stratification-independent baroclinic torque,  $\mathcal{T}_5^*$  represents baroclinic torque  
582 associated with the background pressure field,  $\mathcal{T}_6^*$  represents baroclinic torque associated with the  
583 background specific volume (or density) field, and  $\mathcal{T}_7^*$  represents diffusion associated with variable  
584 viscosity. In the limit as  $M \rightarrow 0$ , both  $\mathcal{T}_6^*$  and  $\mathcal{T}_7^*$  terms go to zero. In the following, we examine  
585 each of these terms to understand their relative effects on the creation and destruction of vorticity  
586 as a function of initial stratification strength. It should be noted that, by focusing this analysis  
587 on the dynamics of the vorticity magnitude  $\omega^*$ , we are able to specifically isolate effects leading to  
588 variations in the strength of vortical motions, independent of the sign of the vorticity.

### 589 C. Effects of Stratification Strength on the Dynamics of the Vorticity

590 Figure 9 shows fields of viscous diffusion,  $\mathcal{T}_2^*$ , dilatation,  $\mathcal{T}_3^*$ , total baroclinic torque,  $\mathcal{T}_{\text{BT}}^* =$   
591  $\mathcal{T}_4^* + \mathcal{T}_5^* + \mathcal{T}_6^*$ , and variable viscosity diffusion,  $\mathcal{T}_7^*$ , for the four different stratification strengths  
592 (with  $Re = 102,000$  in all cases) at a late stage ( $t^* = 20$ ) in the 2D simulations. The vortex  
593 stretching term  $\mathcal{T}_1^*$  is identically zero in 2D and is thus not shown here. Figure 9 shows that, for all  
594 values of  $M$ , the dilatation term  $\mathcal{T}_3^*$  has a similar magnitude to  $\mathcal{T}_{\text{BT}}^*$ , while the constant viscosity  
595 diffusion term,  $\mathcal{T}_2^*$ , is much larger than the variable viscosity contribution,  $\mathcal{T}_7^*$ , and reaches peak  
596 magnitudes similar to, but still smaller than,  $\mathcal{T}_{\text{BT}}^*$ .

597 The relative contributions of the perturbation baroclinic torque,  $\mathcal{T}_4^*$ , the baroclinic torque asso-  
598 ciated with the background pressure,  $\mathcal{T}_5^*$ , and the baroclinic torque associated with the background  
599 density,  $\mathcal{T}_6^*$  to the total baroclinic torque  $\mathcal{T}_{\text{BT}}^*$ , are indicated as a function of stratification strength  
600 in Figure 10. For small  $M$ , Figure 10 shows that  $\mathcal{T}_4^*$ , representing the perturbation baroclinic

601 torque, is primarily positive (indicating vorticity production) and roughly an order of magnitude  
 602 larger than  $\mathcal{T}_5^*$  (baroclinic torque due to the background pressure) and  $\mathcal{T}_6^*$  (baroclinic torque due  
 603 to the background specific volume). The stratification independent baroclinic torque,  $\mathcal{T}_4^*$ , is the  
 604 primary contribution to  $\mathcal{T}_{\text{BT}}^*$  for small  $M$ .

605 Taken together, Figures 9 and 10 thus indicate that the primary dynamical effects for low  $M$   
 606 are the perturbation baroclinic torque (i.e.,  $\mathcal{T}_4^*$ ) and constant viscosity diffusion (i.e.,  $\mathcal{T}_2^*$ ), although  
 607 the former dominates the latter, resulting in the growth of the instability for low  $M$ . The relative  
 608 magnitudes of these terms are shown in Figure 11, where the terms  $\mathcal{T}_i^*$  from Eq. (36) are averaged  
 609 over half of the domain along the  $x_2$  direction to give  $\overline{\mathcal{T}}_i^*$  as a function of  $x_1$ , with the average  
 610 defined for an arbitrary quantity  $f$  as

$$\overline{f}(x_1, t) = \frac{2}{\lambda} \int_0^{\lambda/2} f(x_1, x_2, t) dx_2. \quad (37)$$

611 Figure 11 also shows results for the averages of  $\overline{\mathcal{T}}_i^*$  over  $x_1$  for  $x_1 < 0$  and  $x_1 > 0$ . For the weakest  
 612 stratification examined here, Figure 11(a) shows that the enstrophy is created on average due  
 613 almost entirely to the perturbation baroclinic torque. There is only a relatively small enstrophy  
 614 destruction contribution due to the constant viscosity diffusion.

615 Although the perturbation baroclinic torque  $\mathcal{T}_4^*$  can become locally negative due to density  
 616 inversions (i.e., negative density gradients) created by vortical motions, Figures 10 and 11 show that  
 617 this term remains mostly positive for all but the strongest stratification, due to the presence of the  
 618 instability. Nevertheless,  $\mathcal{T}_4^*$  does decrease in magnitude as  $M$  increases and, in particular, Figure  
 619 11(d) shows that this term can contribute to the destruction of vorticity magnitude for sufficiently  
 620 large stratification. This is consistent with Figure 8, which shows that  $\partial v'^*/\partial x_i^*$  approaches zero,  
 621 while  $\partial p'^*/\partial x_i^*$  becomes close to 1, as  $M$  increases. The reduced vorticity production at larger  
 622 stratifications corresponds to the suppression of the instability growth.

623 For all values of  $M$  considered here, Figures 10 and 11 show that  $\mathcal{T}_4^*$  has the largest contribution  
 624 to the total baroclinic torque in general, but, since it decreases with  $M$ , becomes more similar in  
 625 magnitude to the other terms at the largest stratification considered. At  $M = 1.2$ , the vorticity  
 626 production is much smaller, consistent with the overall suppression of the instability. For large  
 627 stratifications, the reduced vorticity magnitude also translates into lower self-propagating velocity  
 628 for the vortex pairs generated at the bubble/spike interface. In turn, this results in part of the  
 629 fresh fluid brought towards the bubble/spike peaks by the induced vortical velocity returning back  
 630 to the mixing layer. Thus, at  $M = 0.9$ , Figure 11(c) shows density inversions (i.e., negative  $\overline{\mathcal{T}}_4^*$  or



631 stabilizing regions) near the edges of the layer, while at  $M = 1.2$  in Figure 11(d) these regions can  
 632 occur throughout the layer. The reduced self-propagating velocity in the stronger stratification  
 633 cases also appears to be connected to the slower development of the dynamics, which is connected  
 634 to the delayed onset of the RTI suppression shown in Figure 3(b).

635 Variations in the magnitudes of  $\mathcal{T}_5^*$  and  $\mathcal{T}_6^*$  with  $M$  shown in Figures 10 and 11 are somewhat  
 636 more complicated. In particular, the peak magnitudes of both terms increase from  $M = 0.3$ , but  
 637 start decreasing again at larger  $M$  and become smaller for  $M = 1.2$ . Term  $\mathcal{T}_5^*$  reaches its peak  
 638 magnitude at slightly smaller Mach number than  $\mathcal{T}_6^*$  ( $M \sim 0.6$  versus  $M \sim 0.9$ ). At large  $M$ ,  
 639 the gradient contributions to both terms become uniform, with values of 0 and 1, respectively (see  
 640 Figure 8). The prefactors  $(M^2 p_M^* - 1)$  and  $M^2 v_M^*$  depend on the initial background stratification  
 641 and become large in the far-field, but are small near the centerline, even for large  $M$ . Therefore,  
 642 as the instability growth is suppressed at large stratifications,  $\mathcal{T}_5^*$  and  $\mathcal{T}_6^*$  are more confined to the  
 643 region close to the centerline and never reach regions with large prefactor values.

644 Perhaps most significantly, both  $\mathcal{T}_5^*$  and  $\mathcal{T}_6^*$  exhibit asymmetries that affect the overall growth  
 645 of the instability. Aside from local inversions, Figures 10 and 11 show that  $\mathcal{T}_5^*$  presents a top-  
 646 bottom asymmetry with respect to the  $x_1 = 0$  initial location of the instability (i.e., it is positive  
 647 on the spike side and negative on the bubble side). On the other hand, Figure 10 shows that  $\mathcal{T}_6^*$   
 648 presents a left-right asymmetry with respect to the interface between the heavy and light fluid,  
 649 with negative values inside the spike and positive values inside the bubble regions. Conversely, the  
 650 dilatation term  $\mathcal{T}_3^*$  shows the opposite left-right asymmetry, with positive values inside the spike  
 651 and negative values inside the bubble regions. It should be noted, however, that  $\mathcal{T}_5^*$  becomes *larger*  
 652 with respect to  $\mathcal{T}_4^*$  as  $M$  increases. The term  $\mathcal{T}_4^*$  is itself also asymmetric, as shown in Figure 11,  
 653 and it is likely that this asymmetry is the underlying cause of the differences in bubble and spike  
 654 growth rates, particularly for small  $M$ . Moreover, weak asymmetry in bubble and spike growth  
 655 rates is observed in the incompressible limit (i.e.,  $M \rightarrow 0$ ), indicating that  $\mathcal{T}_5^*$  may be a contributor  
 656 to, but not the sole cause of, the asymmetry, since this term approaches zero as  $M \rightarrow 0$ .

657 As explained above, the bubble-spike asymmetry is small in the incompressible case before the  
 658 chaotic stage, it becomes noticeable at  $M = 0.3$ , and then decreases again at large stratifications.  
 659 The history of the top-down asymmetry in the vorticity generation can also be seen from the time  
 660 evolutions of  $\overline{\mathcal{T}}_i$  in Figure 12. This figure does not identify the left-right asymmetry, which has a  
 661 more dynamical effect, as it influences the vortical motions separately within the bubble and spike  
 662 regions. However, it does show that  $\overline{\mathcal{T}}_4^*$  begins symmetrical and develops the top-down asymmetry  
 663 at some later time. On the other hand,  $\overline{\mathcal{T}}_5^*$  is asymmetric from the beginning, such that it represents

664 the source of this asymmetry. This is consistent with the incompressible (i.e.,  $M \rightarrow 0$ ) flow results,  
 665 where terms  $\overline{\mathcal{T}}_3^*$ ,  $\overline{\mathcal{T}}_5^*$ , and  $\overline{\mathcal{T}}_6^*$  are zero, and  $\overline{\mathcal{T}}_4^*$  remains relatively symmetrical until later times.  
 666 Again, at large stratifications, the overall reduction in vorticity production and suppression of the  
 667 instability prevents the bubble/spike asymmetry from becoming more pronounced.

668 Figure 13 shows the time evolution of averages of  $\mathcal{T}_i$ , where the averaging is performed from 0  
 669 to  $\lambda/2$  along the  $x_2$  direction and separately along  $x_1$  from 0 to  $\lambda$ , denoted  $\overline{\overline{\mathcal{T}}_i^+}$ , and from  $-\lambda$  to 0,  
 670 denoted  $\overline{\overline{\mathcal{T}}_i^-}$ . These averaging operators are defined for an arbitrary quantity  $f$  as

$$\overline{\overline{f}}^+ = \frac{2}{\lambda} \int_0^{\lambda/2} \left[ \frac{1}{\lambda} \int_0^\lambda f(x_1, x_2, t) dx_1 \right] dx_2, \quad (38)$$

$$\overline{\overline{f}}^- = \frac{2}{\lambda} \int_0^{\lambda/2} \left[ \frac{1}{\lambda} \int_{-\lambda}^0 f(x_1, x_2, t) dx_1 \right] dx_2. \quad (39)$$

671 As the stratification strength increases, the viscous diffusion and other baroclinic torque terms  
 672 become larger relative to the perturbation baroclinic torque,  $\mathcal{T}_4$ , and the average of  $\mathcal{T}_4$  actually  
 673 begins to decrease at increasingly early times. Figure 13 also indicates that  $\mathcal{T}_4$  is, on average,  
 674 larger for  $x_1/\lambda < 0$  for all times, but this reverses, with  $\mathcal{T}_4$  larger for  $x_1/\lambda > 0$ , as the stratification  
 675 strength increases. As  $M$  increases and the relative magnitude of  $\mathcal{T}_4$  decreases, it is the baroclinic  
 676 torque associated with the background pressure field (i.e.,  $\mathcal{T}_5$ ) that becomes correspondingly more  
 677 dominant in the overall dynamics. This term is strongly asymmetric and leads to vorticity creation  
 678 for  $x_1/\lambda < 0$  and destruction for  $x_1/\lambda > 0$ .

679 Finally, Figures 9, 11, and 12 show that the variable viscosity diffusion term,  $\mathcal{T}_7^*$ , is negligible for  
 680 all  $M$  considered here and all times. By contrast, the magnitude of the constant viscosity diffusion  
 681 term,  $\mathcal{T}_2^*$ , remains more uniform with increasing  $M$ , although it does become more consistently  
 682 negative as  $M$  increases, as shown most clearly in Figure 12. This indicates that constant viscosity  
 683 diffusion is the primary term leading to destruction of vorticity magnitude, and this term begins  
 684 to rival the magnitude of the perturbation baroclinic torque term (i.e.,  $\mathcal{T}_4^*$ ) for large  $M$ .

685 Taken together, these results are indicative of larger vorticity production within the spike re-  
 686 gion, as compared to the bubble region, due to compressibility and stratification effects. Because  
 687 the bubble and spike vertical axes are maintained throughout the flow evolution for the single  
 688 mode case, the vorticity field itself retains a similar symmetry. This results in an induced vortical  
 689 velocity along the bubble/spike axes, which helps the instability grow, similar to the incompressible  
 690 (i.e.,  $M \rightarrow 0$ ) case [34]. However, for the compressible case, the dilatation term and baroclinic  
 691 contributions sum up to a bubble/spike asymmetry even at low Mach numbers. At higher Mach

692 numbers, due to the overall suppression of the instability, these contributions also decrease and the  
 693 asymmetry becomes small again. Overall, the primary dynamical balance is between constant vis-  
 694 cosity diffusion, which leads to the destruction of vorticity magnitude, and perturbation baroclinic  
 695 torque, which leads to vorticity magnitude production. It should be noted that the asymmetry  
 696 in the overall dynamics is fundamentally attributable to the presence of the asymmetric back-  
 697 ground stratification, where the magnitude of the background pressure gradient is larger above  
 698 the initial interface at  $x_1/\lambda = 0$  than below the interface. If the background stratification were  
 699 instead uniform (i.e., a linear variation in background pressure), then the asymmetry observed in  
 700 the present study would not be expected to form. Consequently, the present asymmetry should  
 701 not be considered a non-Boussinesq effect.

## 702 VI. CONCLUSIONS

703 In the present study, wavelet-based adaptive mesh refinement has been used to perform DNS of  
 704 2D single-mode compressible low Atwood number RTI for four different isothermal stratification  
 705 strengths, corresponding to Mach numbers from 0.3 to 1.2, and for three different perturbation  
 706 Reynolds numbers from 25,500 to 102,000. The simulation results have been examined to under-  
 707 stand the effects of stratification strength and Reynolds number on the characteristics, dynamics,  
 708 and rate of RTI growth. In the present context, compressibility is controlled through the values  
 709 of the background pressure at the interface between the heavier and lighter fluids, which also af-  
 710 fects the background stratification strength, and would be considered flow, as opposed to fluid,  
 711 compressibility. In this context, the incompressible limit (i.e.,  $M \rightarrow 0$ ) is reached as the speed of  
 712 sounds goes to infinity by increasing the interface pressure and temperature, such that the inter-  
 713 face density remains constant. The practical setup corresponds to an enclosed fluid system that is  
 714 uniformly heated (i.e., heating at constant volume).

715 For weak stratifications, RTI growth was found to undergo a re-acceleration after reaching a  
 716 plateau in the growth rate that approximately matched predictions from potential flow theory. As  
 717 the stratification strength increased, however, this re-acceleration was found to no longer occur,  
 718 and the RTI growth was suppressed; this suppression occurred in the present study for all Mach  
 719 numbers greater than 0.3. For weak stratifications, the bubble was found to grow at a slower rate  
 720 than the spike, but this asymmetry progressively weakened as the stratification strength increased.  
 721 The Reynolds number was found to have little impact on RTI growth for the range of Mach numbers  
 722 and for the simulation length examined here. However, small-scale structure was found to become

723 more pronounced as the Reynolds number increased. At very early times, during the diffusive  
724 stage, the growth rates were larger at smaller Reynolds numbers, but the instability became faster  
725 during the linear and weakly nonlinear stages at higher Reynolds numbers, consistent with prior  
726 studies of Reynolds number effects [15, 34].

727 To determine the origins of the observed results, the dynamics of the vorticity magnitude were  
728 examined in detail. A non-dimensional compressible vorticity transport equation was derived to  
729 explicitly show dependencies on the Mach, Atwood, and Reynolds numbers, and the effects of  
730 stratification strength were studied for each of the terms in the transport equation. This analysis  
731 showed that incompressible baroclinic torque was the dominant driver of RTI growth for the range  
732 of stratifications considered, and its decrease at higher stratifications corresponded to the overall  
733 instability suppression. Asymmetries in the RTI growth were found to be the result of compress-  
734 ibility effects, as a consequence of the dilatation term and background stratification contributions  
735 to the baroclinic torque. However, for strong stratifications, since the instability did not evolve far  
736 from the centerline, the latter contributions remain small and the bubble/spike asymmetry does  
737 not become pronounced.

738 In total, the simulations and analysis performed in this study have enabled the three questions  
739 posed in Section I to be fully addressed. However, much work remains to be done. In particular, the  
740 present analysis of vorticity dynamics should be extended to multi-mode initial perturbations, to  
741 different stratification types (e.g., isopycnic and isentropic stratifications), and to 3D cases where  
742 vortex stretching effects in the vorticity dynamics are nonzero. It would also be of interest to  
743 explore longer simulation times for the weakly stratified cases to determine whether the chaotic  
744 development regime noted by Wei & Livescu [34] is recovered in the context of fully compressible  
745 simulations.

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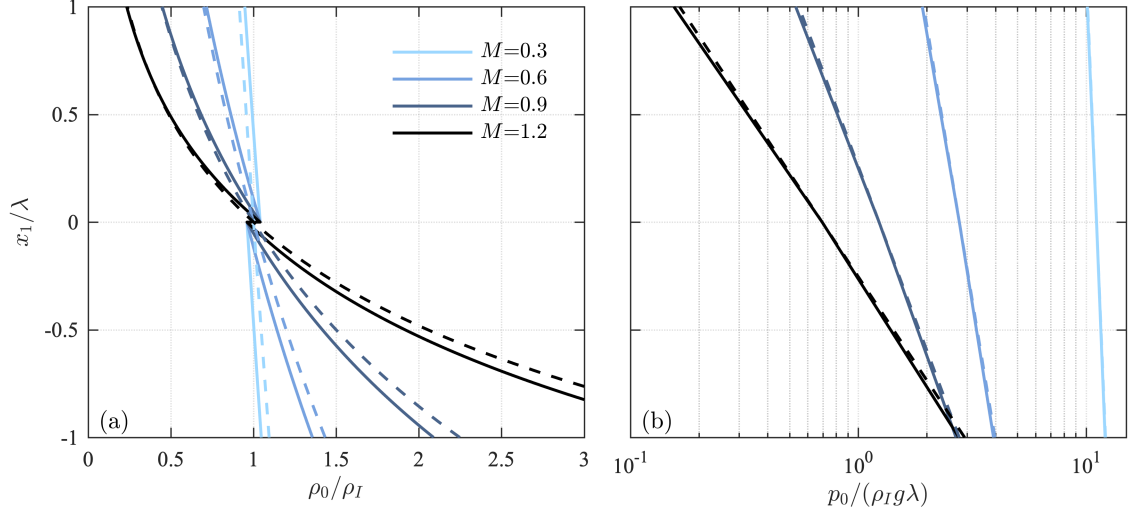


FIG. 1. [Color online] Background density (a) and pressure (b) profiles for  $A = 0.04$  and stratification strengths from  $M = 0.3$  to 1.2. The background states, indicated by solid lines, are hydrostatic and are given by Eqs. (23) and (24). The density difference at  $x_1 = 0$  is determined by  $A$ . The dashed lines show the  $A = 0$  background profiles used in Section V for the analysis of baroclinic torque in the vorticity equation.

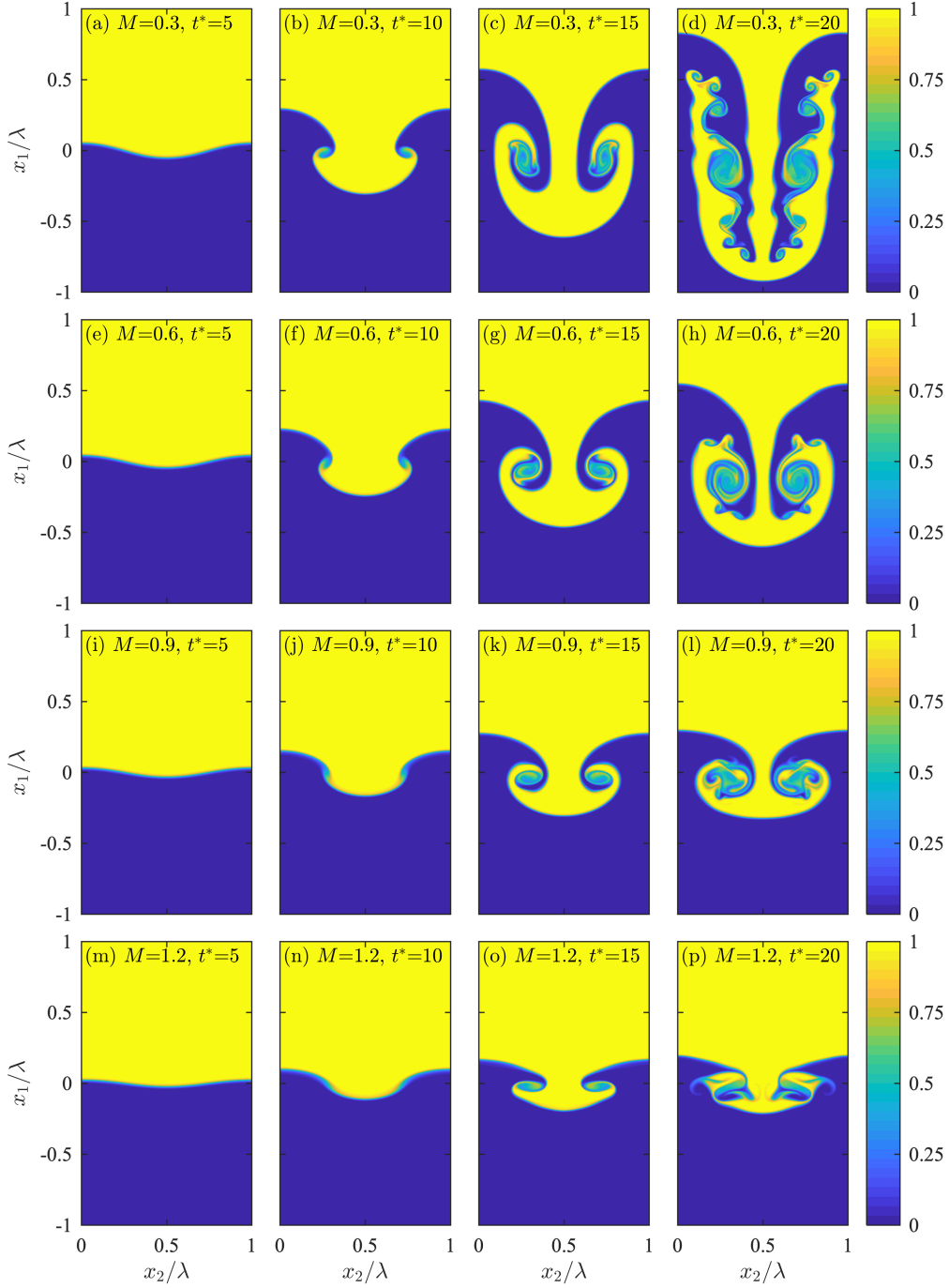


FIG. 2. [Color online] Instantaneous fields of the heavier species mass fraction,  $Y_2$ , in  $x_1$ - $x_2$  planes as a function of non-dimensional time  $t^* = t\sqrt{g/\lambda}$  for stratification strengths  $M = 0.3, 0.6, 0.9,$  and  $1.2$  (increasing from top to bottom). The progression in time from  $t^* = 5$  to  $t^* = 20$  is shown in columns from left to right.

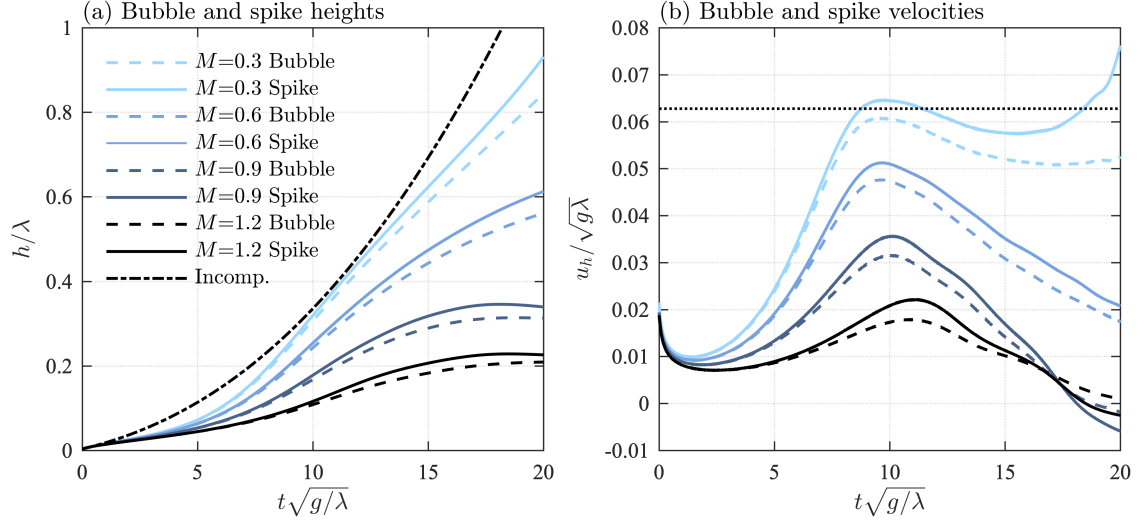


FIG. 3. [Color online] Time series of bubble and spike tip heights,  $h$ , (panel a) and velocities,  $u_h$ , (panel b) for  $M = 0.3, 0.6, 0.9$ , and  $1.2$ . Bubble results are shown by dashed lines and spike results are shown by solid lines. Heights, velocities, and times have each been non-dimensionalized using  $\lambda$  and  $g$ . The dash-dot line in panel (a) shows incompressible results from Wei & Livescu [34] and the horizontal dotted line in panel (b) shows the predicted bubble velocity from drag and potential flow models [50–52],  $u_h/\sqrt{g\lambda} \approx 0.063$ .

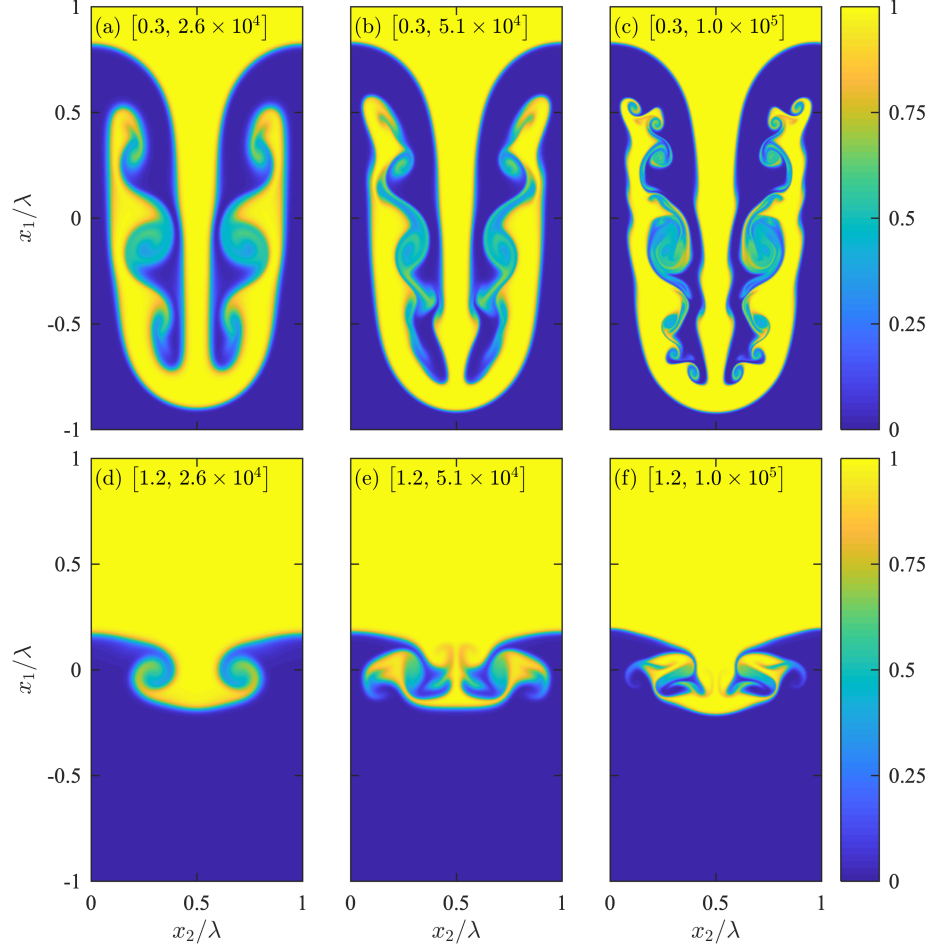


FIG. 4. [Color online] Instantaneous fields of the heavier species mass fraction,  $Y_2$ , in  $x_1$ - $x_2$  planes for Reynolds numbers  $Re = 2.55 \times 10^4$ ,  $5.10 \times 10^4$ , and  $1.02 \times 10^5$  (left to right columns), for stratification strengths  $M = 0.3$  (top row) and  $M = 1.2$  (bottom row). Panel labels are defined as  $[M, Re]$ . Results are shown at  $t^* = t\sqrt{g/\lambda} = 20$  in each case.

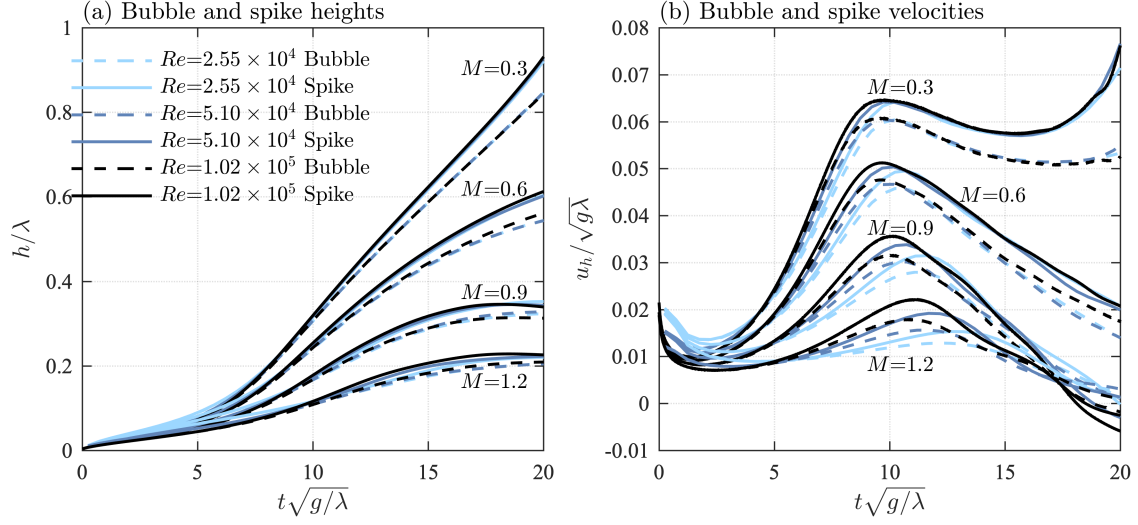


FIG. 5. [Color online] Time series of bubble and spike tip heights,  $h$ , (panel a) and velocities,  $u_h$ , (panel b) for  $M = 0.3, 0.6, 0.9$ , and  $1.2$ , at  $Re = 2.55 \times 10^4, 5.10 \times 10^4$ , and  $1.02 \times 10^5$ . Bubble results are shown by dashed lines and spike results are shown by solid lines. Results for  $Re = 2.55 \times 10^4, 5.10 \times 10^4$ , and  $1.02 \times 10^5$  are shown using black, red, and blue lines, respectively. Heights, velocities, and times have each been non-dimensionalized using  $\lambda$  and  $g$ .

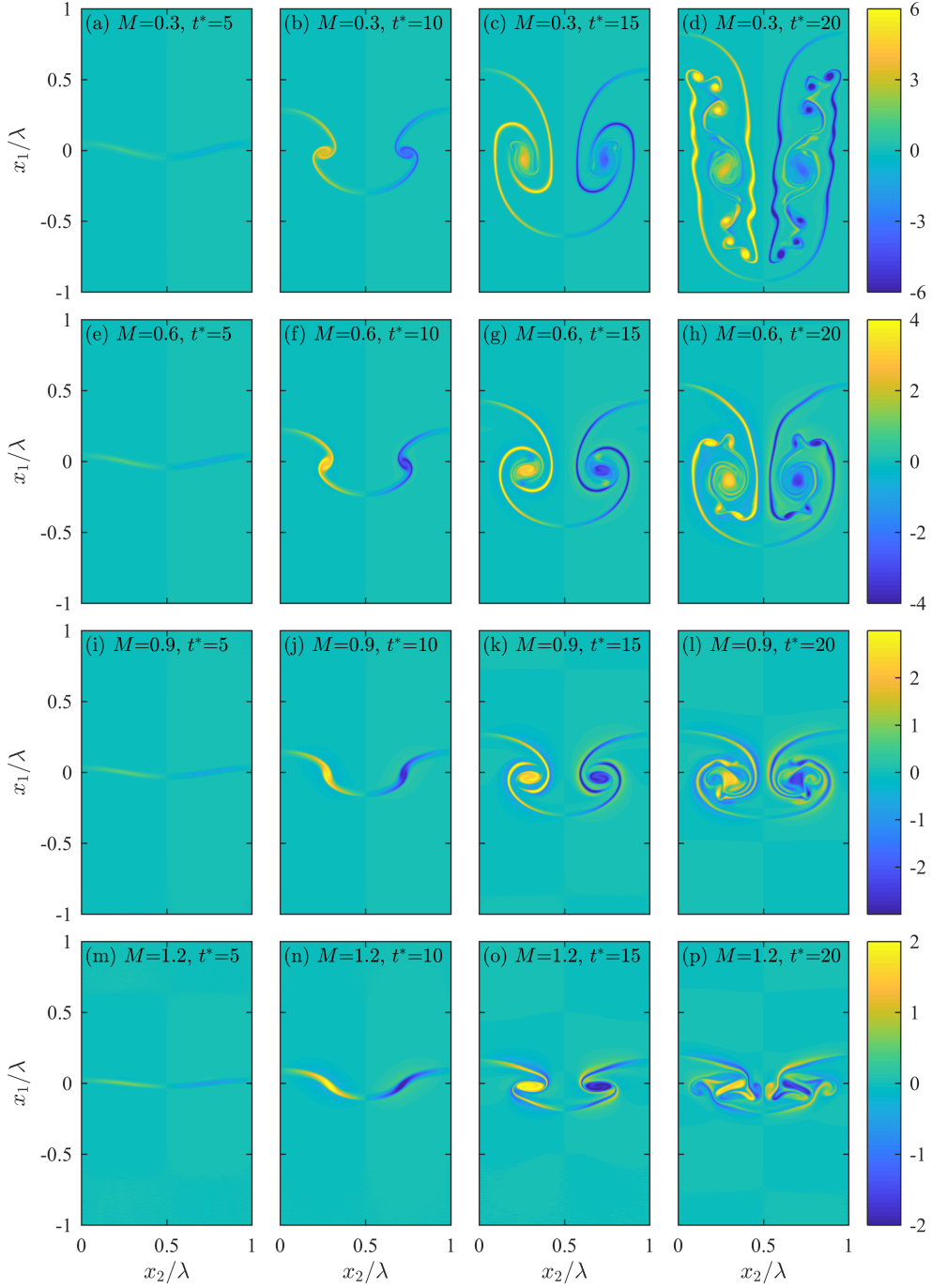


FIG. 6. [Color online] Instantaneous fields of the non-dimensional vorticity  $\omega_3^* = \omega_3 \sqrt{\lambda/g}$ , in  $x_1$ - $x_2$  planes as a function of non-dimensional time  $t^* = t\sqrt{Ag/\lambda}$  for stratification strengths  $M = 0.3, 0.6, 0.9,$  and  $1.2$  (increasing from top to bottom). The progression in time from  $t^* = 1$  to  $t^* = 4$  is shown in columns from left to right.

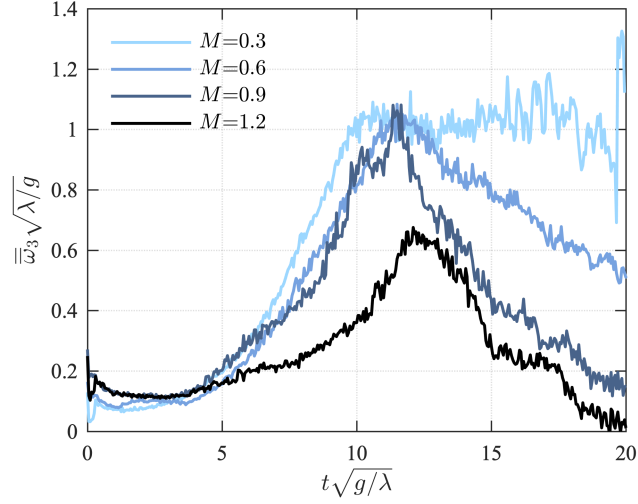


FIG. 7. [Color online] Temporal evolution of the average vorticity  $\overline{\omega}_3$  over the left half of the domain (i.e.,  $x_2 < \lambda/2$ ) for stratification strengths  $M = 0.3, 0.6, 0.9,$  and  $1.2$ , where the averaging operator is defined in Eq. (26).

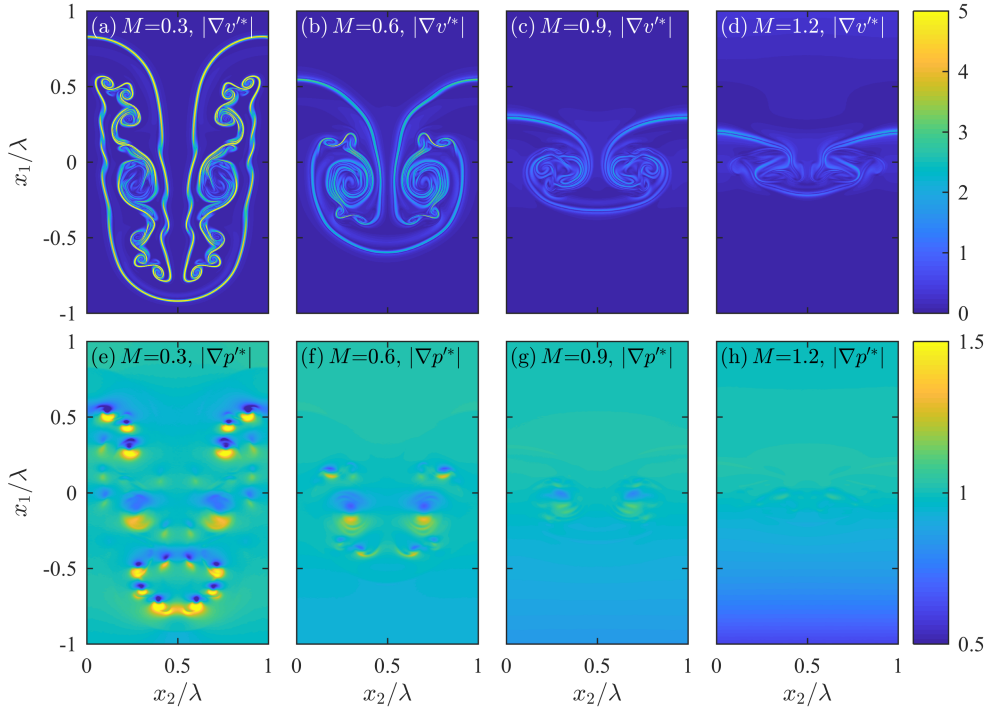


FIG. 8. [Color online] Instantaneous fields showing the magnitudes of  $\partial v^{*}/\partial x_i^{*}$  (top row) and  $\partial p^{*}/\partial x_i^{*}$  (bottom row), where the perturbation gradients are given in Eqs. (32) and (33). Fields are shown at non-dimensional time  $t^{*} = t\sqrt{Ag/\lambda} = 20$  and for stratification strengths  $M = 0.3, 0.6, 0.9,$  and  $1.2$  (left to right).

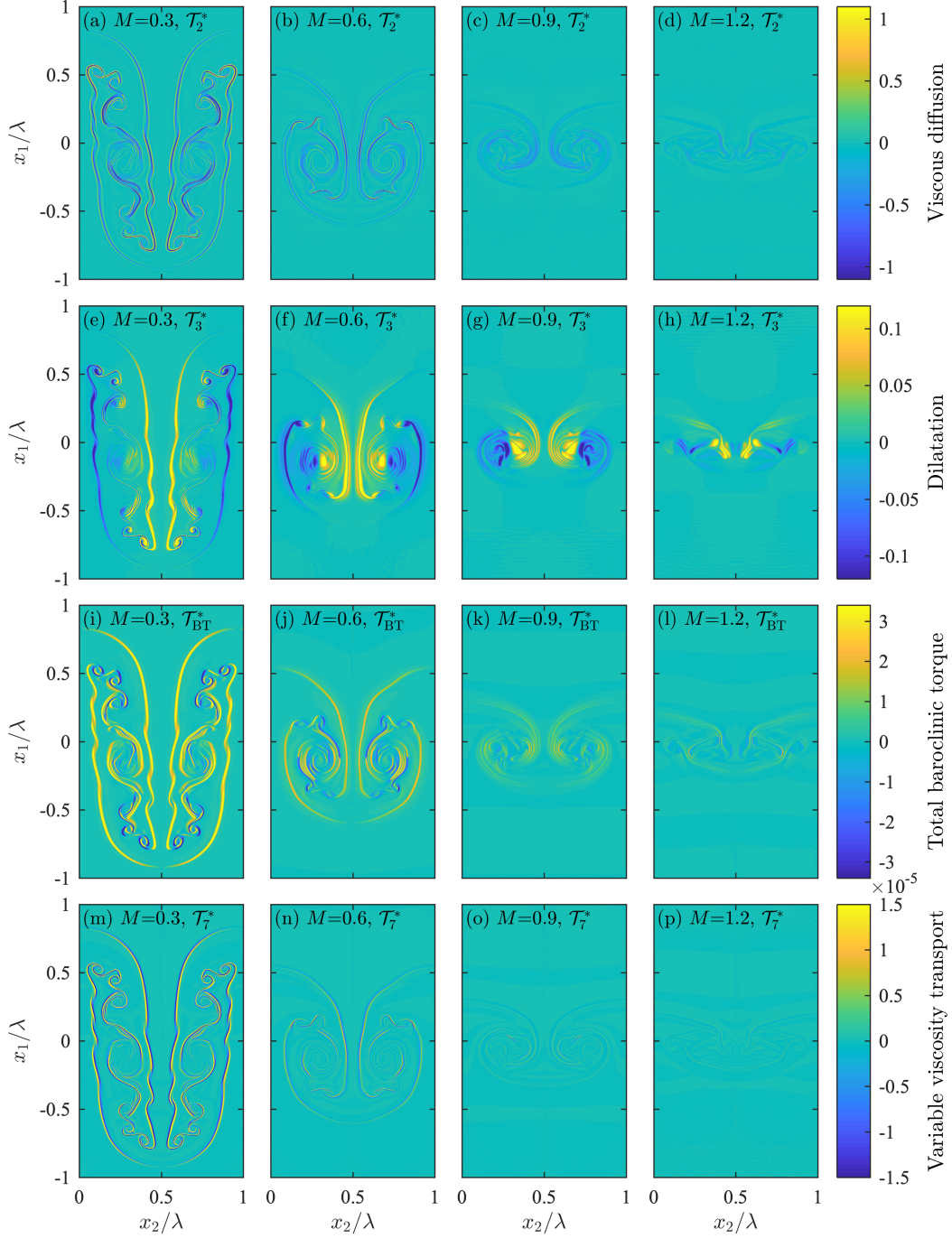


FIG. 9. [Color online] Instantaneous fields of  $\mathcal{T}_i^* = \mathcal{T}_i(\lambda/g)$  appearing in Eq. (36), which describes the dynamics of  $\omega^* = |\omega_3^*|$ , for the 2D simulation cases. Fields are shown at non-dimensional time  $t^* = t\sqrt{g/\lambda} = 20$  for (from top to bottom)  $\mathcal{T}_2^*$  (viscous diffusion),  $\mathcal{T}_3^*$  (dilatation),  $\mathcal{T}_{BT}^*$  (total baroclinic torque), and  $\mathcal{T}_7^*$  (variable viscosity transport) and for stratification strengths  $M = 0.3, 0.6, 0.9$ , and  $1.2$  (left to right). Note that the color axes are different for each term.



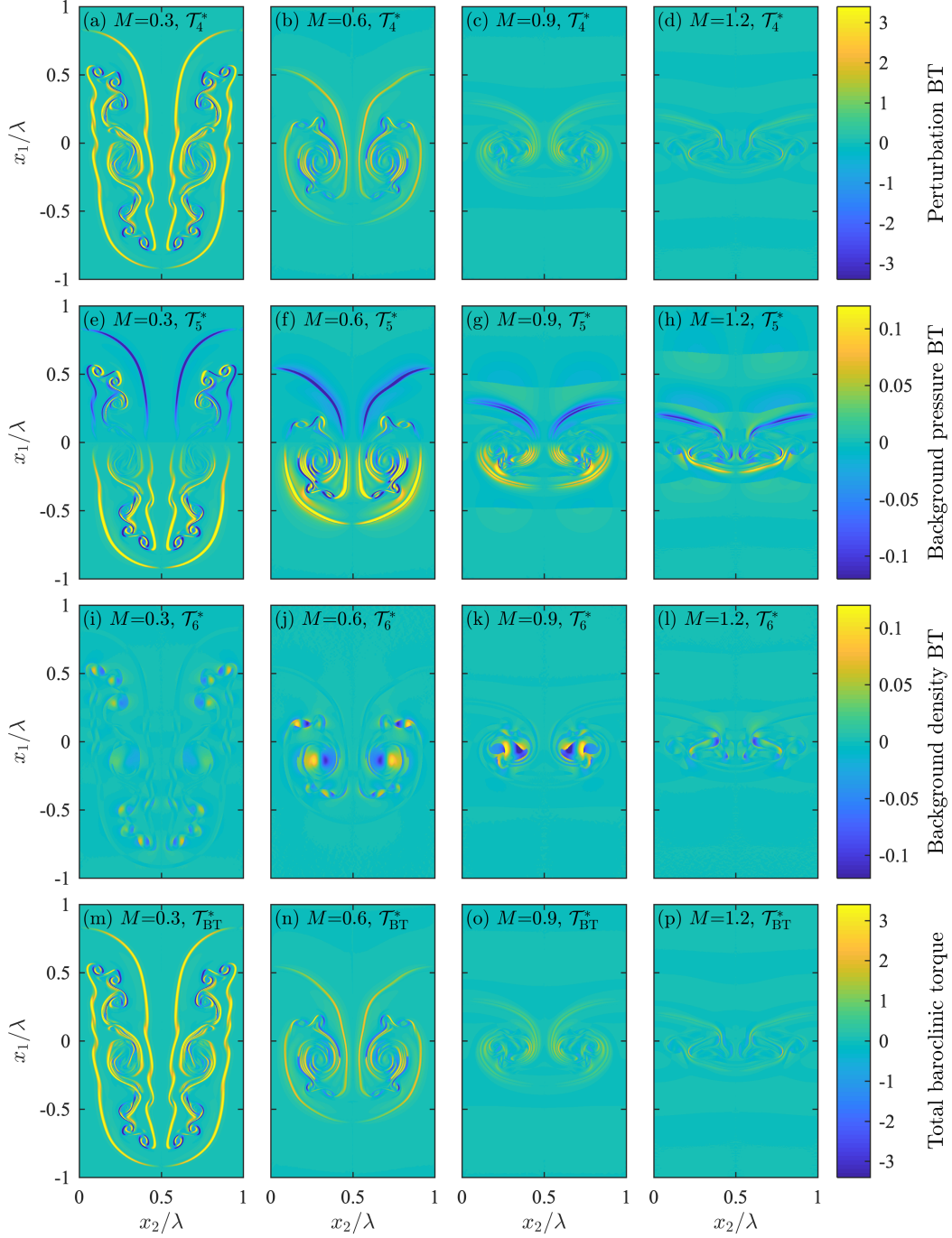


FIG. 10. [Color online] Instantaneous fields of the baroclinic torque terms appearing in Eq. (36), which describes the dynamics of  $\omega^* = |\omega_3^*|$ , for the 2D simulation cases. Fields are shown at non-dimensional time  $t^* = t\sqrt{g/\lambda} = 20$  for (from top to bottom)  $\mathcal{T}_4^*$  (perturbation baroclinic torque),  $\mathcal{T}_5^*$  (baroclinic torque associated with the background pressure),  $\mathcal{T}_6^*$  (baroclinic torque associated with the background density), and  $\mathcal{T}_{\text{BT}}^* = \mathcal{T}_4^* + \mathcal{T}_5^* + \mathcal{T}_6^*$  (total baroclinic torque) and for stratification strengths  $M = 0.3, 0.6, 0.9$ , and  $1.2$  (left to right). Note that the color axes are different for each term.

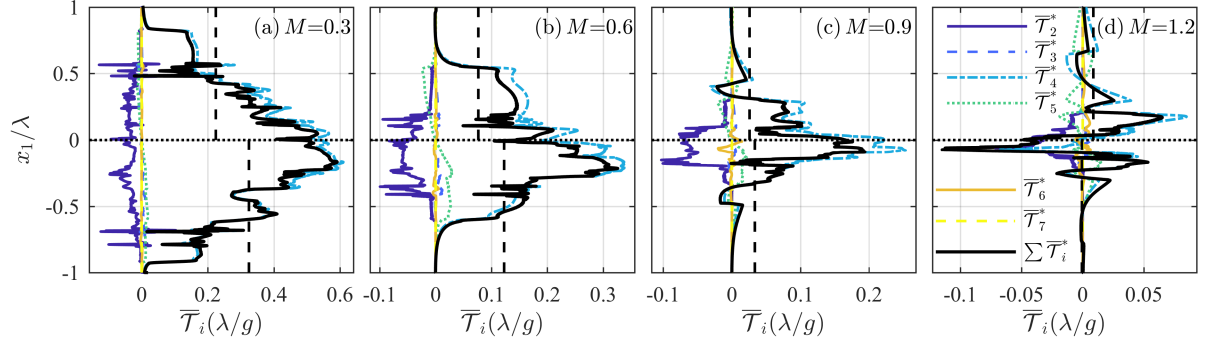


FIG. 11. [Color online] Spatial dependence along the  $x_1$  direction of the half-domain averages of  $\mathcal{T}_2^* - \mathcal{T}_7^*$  appearing in Eq. (36) for stratification strengths  $M = 0.3, 0.6, 0.9,$  and  $1.2$  (a-d) in the 2D simulation cases. The sums of terms  $\mathcal{T}_2^* - \mathcal{T}_7^*$  are also shown. The averaging operator  $\overline{(\cdot)}$  is defined in Eq. (37) and  $\overline{\mathcal{T}}_i$  is written in non-dimensional form as  $\overline{\mathcal{T}}_i^* = \overline{\mathcal{T}}_i(\lambda/g)$ . The vertical black dashed lines show averages of the sum of all terms for  $x_1/\lambda > 0$  and  $x_1/\lambda < 0$ . All results are shown at a non-dimensional time of  $t^* = t\sqrt{g/\lambda} = 20$ .

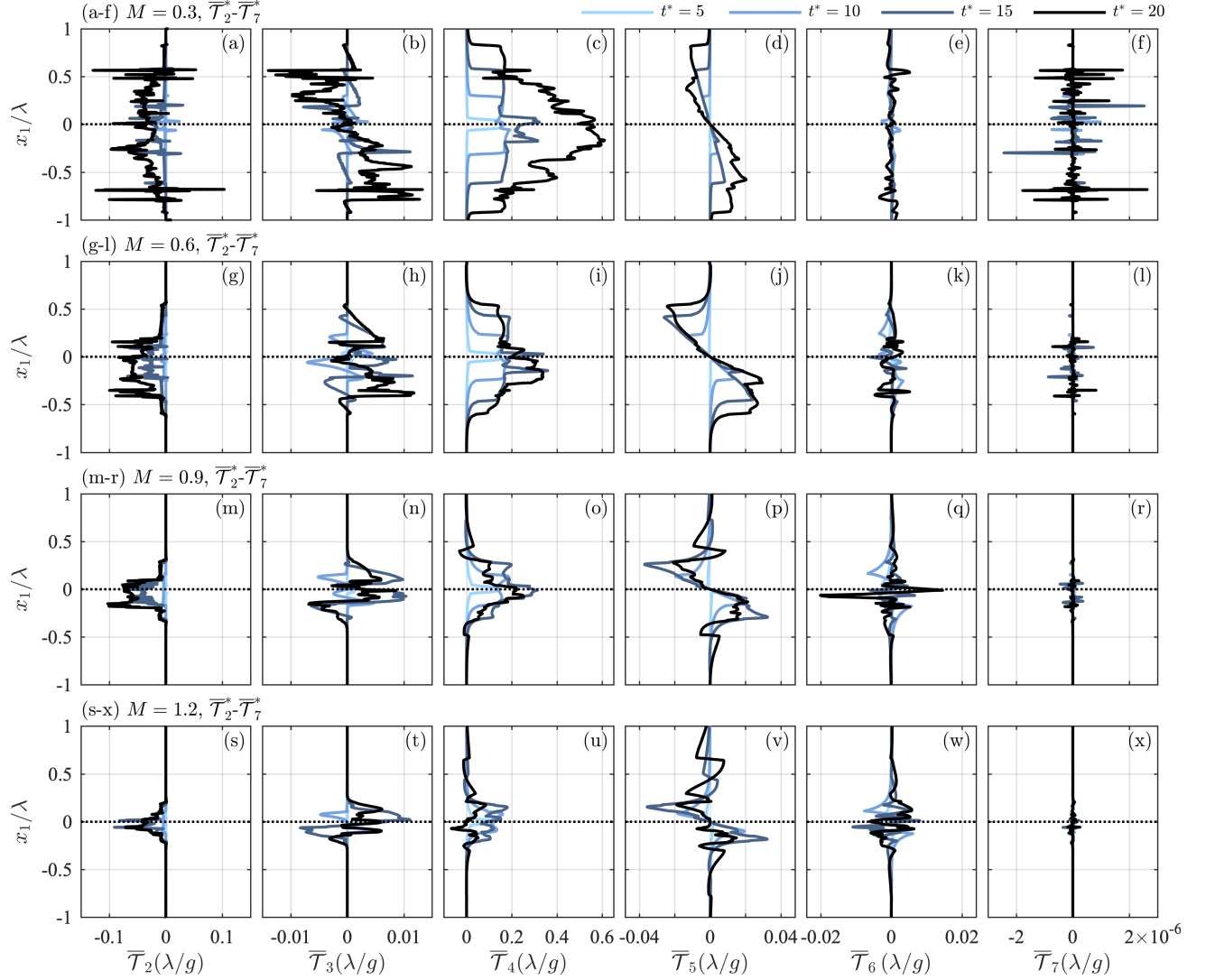


FIG. 12. [Color online] Spatial dependence of the half-domain averages of  $\overline{\mathcal{T}_2^* - \mathcal{T}_7^*}$  (columns from left to right) appearing in Eq. (36) for stratification strengths  $M = 0.3$  (a-f),  $M = 0.6$  (g-l),  $M = 0.9$  (m-r), and  $M = 1.2$  (s-x) at non-dimensional times  $t^* = t\sqrt{g/\lambda} = 5, 10, 15,$  and  $20$ . The averaging operator  $\overline{(\cdot)}$  is defined in Eq. (37) and  $\overline{\mathcal{T}_i}$  is written in non-dimensional form as  $\overline{\mathcal{T}_i^*} = \overline{\mathcal{T}_i}(\lambda/g)$ .

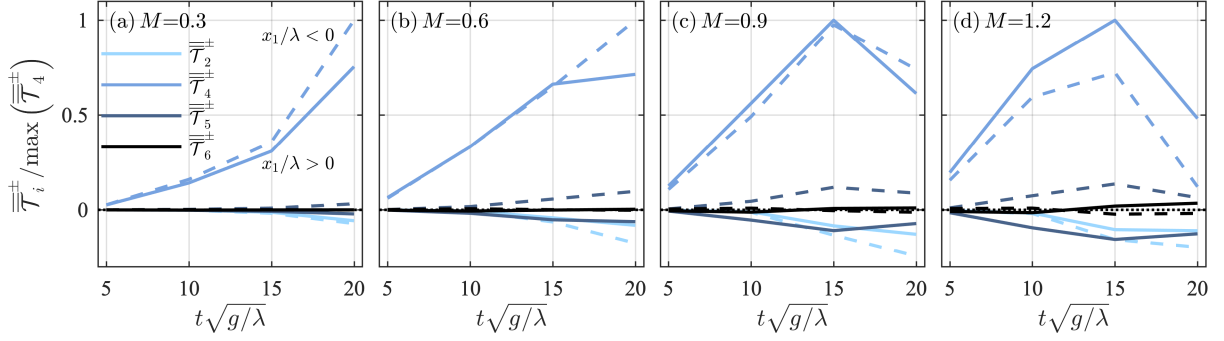


FIG. 13. [Color online] Temporal dependence of half-domain averages of the dilatation,  $\overline{\mathcal{T}}_2$ , perturbation baroclinic torque,  $\overline{\mathcal{T}}_4$ , the baroclinic torque associated with background pressure,  $\overline{\mathcal{T}}_5$ , and the baroclinic torque associated with the background density,  $\overline{\mathcal{T}}_6$  for non-dimensional times  $t^* = t\sqrt{g/\lambda} = 5, 10, 15,$  and  $20$  and for stratification strengths  $M = 0.3, 0.6, 0.9,$  and  $1.2$  (a-d). The half-domain averages used here are defined in Eqs. (38) and (39), with  $\overline{\mathcal{T}}_i^+$  indicating an average over  $x_1/\lambda > 0$  (solid lines) and  $\overline{\mathcal{T}}_i^-$  indicating an average over  $x_1/\lambda < 0$  (dashed lines). All results are normalized by the maximum values of the perturbation baroclinic torque over both halves of the domain for each  $M$ .